Structural Analysis III

The Moment Area Method – Mohr's Theorems

2009/10

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Contents

1. Introduction

1.1 Purpose

The moment-area method, developed by Otto Mohr in 1868, is a powerful tool for finding the deflections of structures primarily subjected to bending. Its ease of finding deflections of determinate structures makes it ideal for solving indeterminate structures, using compatibility of displacement.

Otto C. Mohr (1835-1918)

Mohr's Theorems also provide a relatively easy way to derive many of the classical methods of structural analysis. For example, we will use Mohr's Theorems later to derive the equations used in Moment Distribution. The derivation of Clayperon's Three Moment Theorem also follows readily from application of Mohr's Theorems.

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2. Theory

2.1 Basis

We consider a length of beam *AB* in its undeformed and deformed state, as shown on the next page. Studying this diagram *carefully*, we note:

1. *AB* is the original unloaded length of the beam and *A'B'* is the deflected position of *AB* when loaded.

2. The angle subtended at the centre of the arc *A'OB'* is θ and is the change in curvature from *A'* to *B'*.

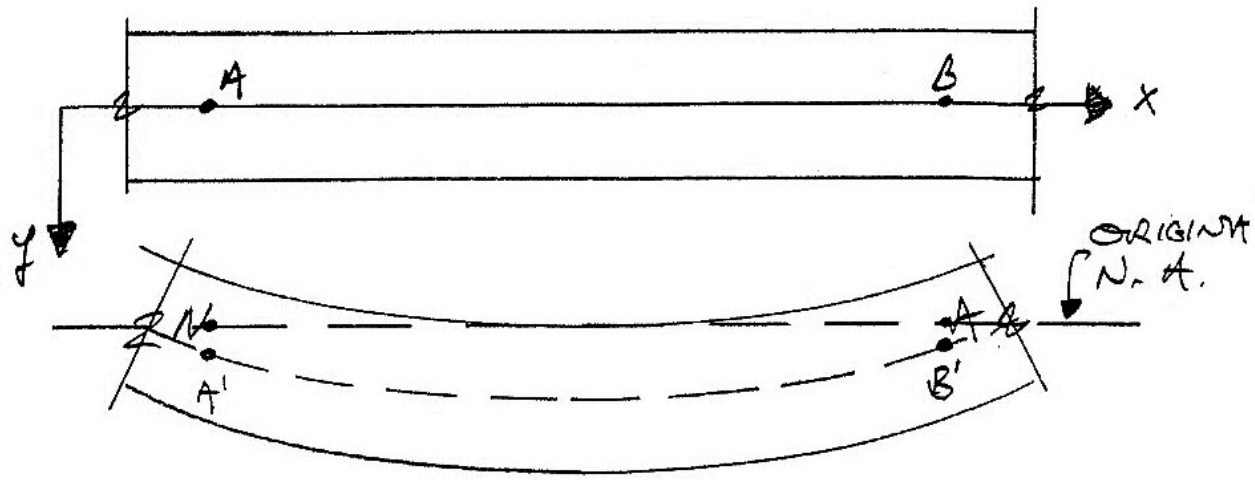
3. *PQ* is a very short length of the beam, measured as ds along the curve and dx along the *x*-axis.

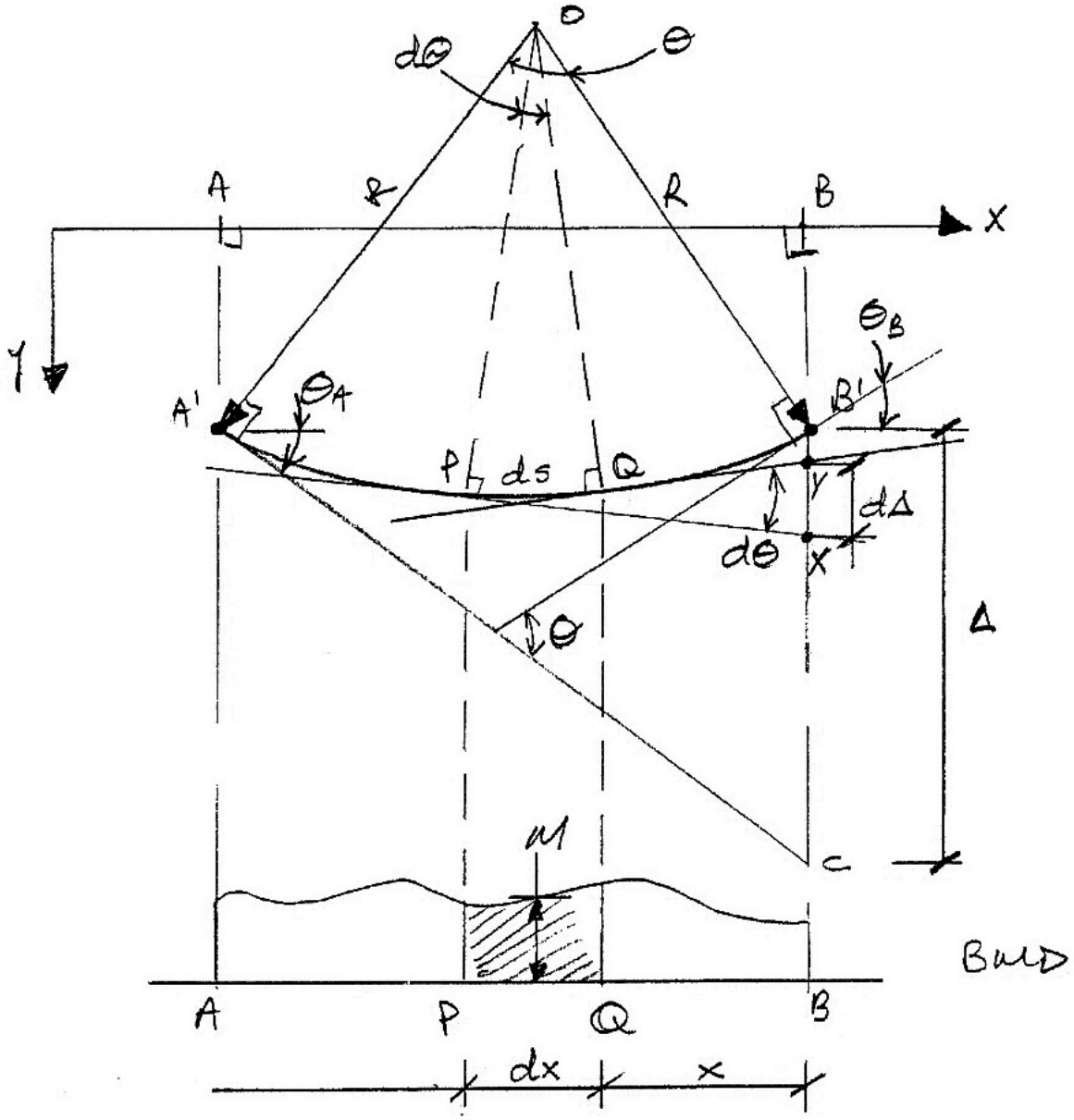
4. $d\theta$ is the angle subtended at the centre of the arc ds.

5. $d\theta$ is the change in curvature from *P* to *Q*.

6. *M* is the average bending moment over the portion dx between *P* and *Q*.

7. The distance Δ is known as the vertical intercept and is the distance from *B'* to the produced tangent to the curve at *A'* which crosses under *B'* at *C*. It is measured perpendicular to the undeformed neutral axis (i.e. the *x*-axis) and so is 'vertical'.





Basis of Theory

2.2 Mohr's First Theorem (Mohr I)

Development

Noting that the angles are always measured in radians, we have:

$$ds = R \cdot d\theta$$
$$\therefore R = \frac{ds}{d\theta}$$

From the Euler-Bernoulli Theory of Bending, we know:

$$\frac{1}{R} = \frac{M}{EI}$$

Hence:

$$d\theta = \frac{M}{EI} \cdot ds$$

But for small deflections, the chord and arc length are similar, i.e. $ds \approx dx$, giving:

$$d\theta = \frac{M}{EI} \cdot dx$$

The total change in rotation between A and B is thus:

$$\int_A^B d\theta = \int_A^B \frac{M}{EI}\, dx$$

The term M/EI is the curvature and the diagram of this term as it changes along a beam is the curvature diagram (or more simply the M/EI diagram). Thus we have:

$$d\theta_{BA} = \theta_B - \theta_A = \int_A^B \frac{M}{EI}\, dx$$

This is interpreted as:

$$\boxed{\left[\text{Change in slope}\right]_{AB} = \left[\text{Area of } \frac{M}{EI} \text{ diagram}\right]_{AB}}$$

This is Mohr's First Theorem (Mohr I):

> *The change in slope over any length of a member subjected to bending is equal to the area of the curvature diagram over that length.*

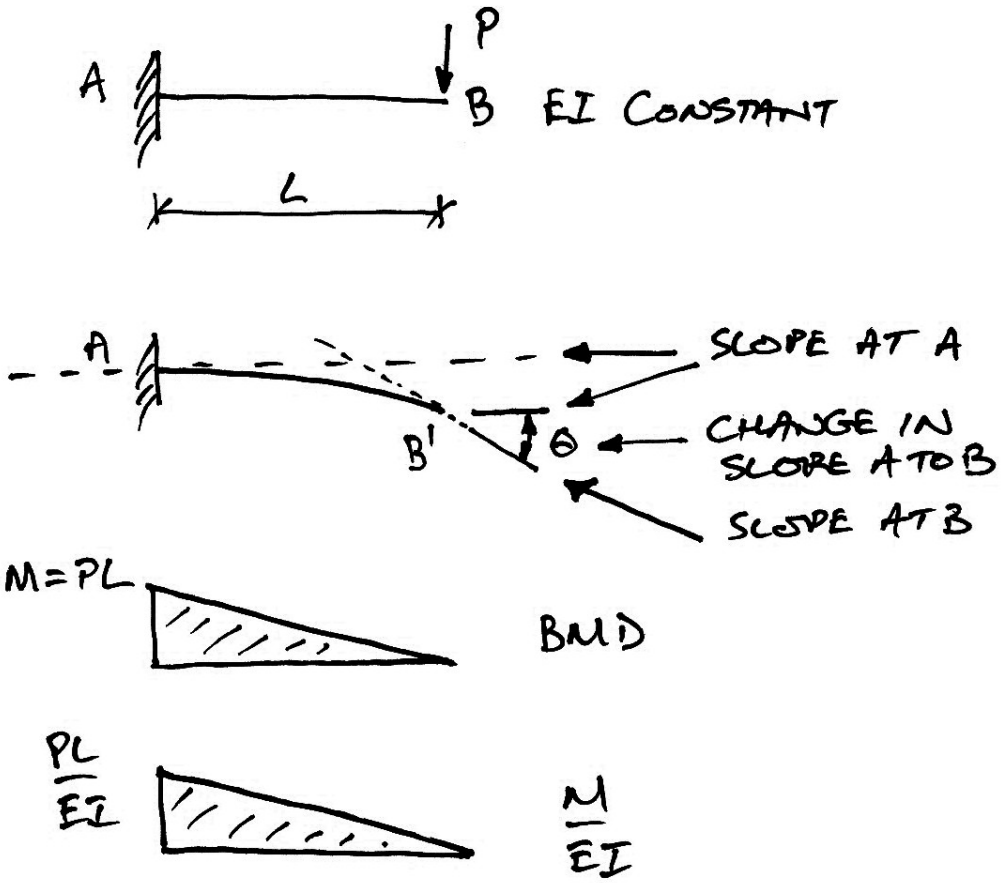
Usually the beam is prismatic and so E and I do not change over the length AB, whereas the bending moment M will change. Thus:

$$\theta_{AB} = \frac{1}{EI}\int_A^B M\ dx$$

$$\left[\text{Change in slope}\right]_{AB} = \frac{\left[\text{Area of } M \text{ diagram}\right]_{AB}}{EI}$$

Example 1

For the cantilever beam shown, we can find the rotation at *B* easily:



Thus, from Mohr I, we have:

$$[\text{Change in slope}]_{AB} = \left[\text{Area of } \frac{M}{EI} \text{ diagram}\right]_{AB}$$

$$\theta_B - \theta_A = \frac{1}{2} \cdot L \cdot \frac{PL}{EI}$$

Since the rotation at *A* is zero (it is a fixed support), i.e. $\theta_A = 0$, we have:

$$\theta_B = \frac{PL^2}{2EI}$$

2.3 Mohr's Second Theorem (Mohr II)

Development

From the main diagram, we can see that:

$$d\Delta = x \cdot d\theta$$

But, as we know from previous,

$$d\theta = \frac{M}{EI} \cdot dx$$

Thus:

$$d\Delta = \frac{M}{EI} \cdot x \cdot dx$$

And so for the portion *AB*, we have:

$$\int_A^B d\Delta = \int_A^B \frac{M}{EI} \cdot x \cdot dx$$

$$\Delta_{BA} = \left[\int_A^B \frac{M}{EI} \cdot dx \right] \overline{x}$$

$$= \text{ First moment of } \frac{M}{EI} \text{ diagram about } B$$

This is easily interpreted as:

$$\begin{bmatrix}\text{Vertical}\\ \text{Intercept}\end{bmatrix}_{BA} = \begin{bmatrix}\text{Area of}\\ \dfrac{M}{EI}\text{ diagram}\end{bmatrix}_{BA} \times \begin{bmatrix}\text{Distance from } B \text{ to centroid}\\ \text{of } \left(\dfrac{M}{EI}\right)_{BA}\text{ diagram}\end{bmatrix}$$

This is Mohr's Second Theorem (Mohr II):

For an originally straight beam, subject to bending moment, the vertical intercept between one terminal and the tangent to the curve of another terminal is the first moment of the curvature diagram about the terminal where the intercept is measured.

There are two crucial things to note from this definition:

- Vertical intercept is *not* deflection; look again at the fundamental diagram – it is the distance from the deformed position of the beam to the tangent of the deformed shape of the beam at another location. That is:

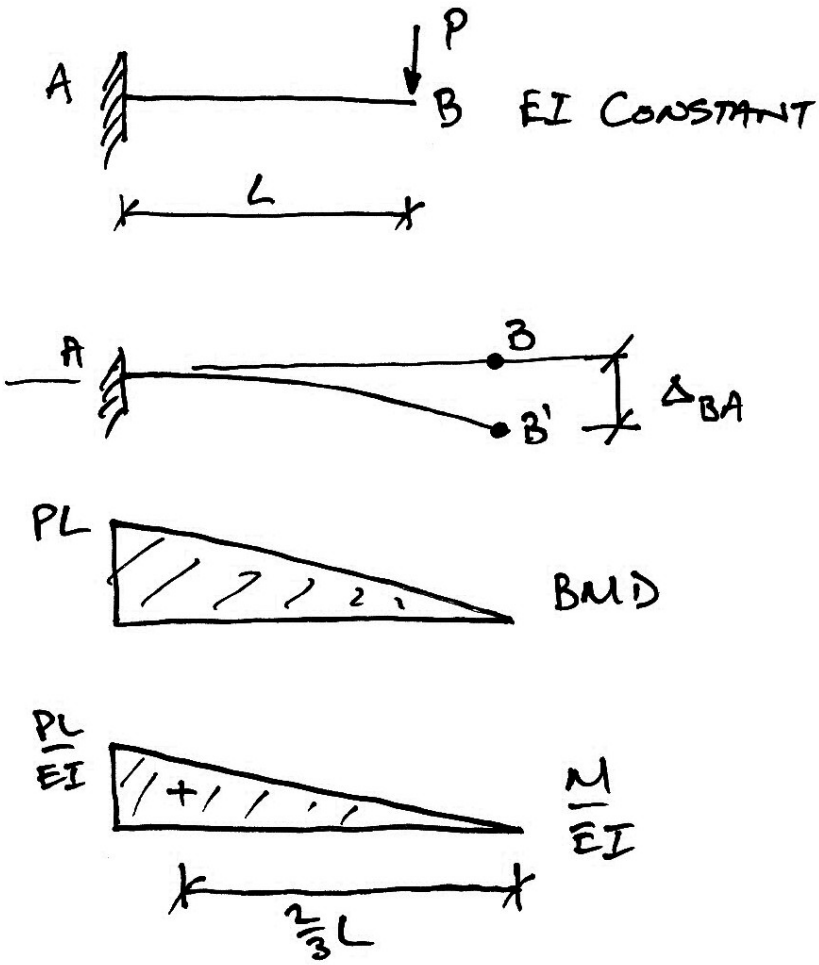
$$\Delta \neq \delta$$

- The moment of the curvature diagram must be taken about the point where the vertical intercept is required. That is:

$$\Delta_{BA} \neq \Delta_{AB}$$

Example 2

For the cantilever beam, we can find the defection at *B* since the produced tangent at *A* is horizontal, i.e. $\theta_A = 0$. Thus it can be used to measure deflections from:



Thus, from Mohr II, we have:

$$\Delta_{BA} = \left[\frac{1}{2} \cdot L \cdot \frac{PL}{EI}\right]\left[\frac{2L}{3}\right]$$

And so the deflection at *B* is:

$$\delta_B = \frac{PL^3}{3EI}$$

2.4 Sign Convention

The sign convention for application of Mohr's Theorems is unfortunately not easily remembered. It is usually easier to note the sense of displacements instead. However, the sign convention is as follows.

Mohr's First Theorem

- If the net area of the BMD is positive (mostly sagging), the change in rotation between two points is measured *anti-clockwise* from the tangent of the first point.

- Conversely, if the net BMD area is negative (mostly hogging), the change in rotation is measured *clockwise* from the tangent to the first point.

Sketch these:

Mohr's Second Theorem

- If the net moment of area of the BMD is positive (mostly sagging), the deflected position of the member lies *above* the produced tangent.

- Conversely, if the net moment of area of the BMD area is negative (mostly hogging), the deflected position of the member lies *below* the produced tangent.

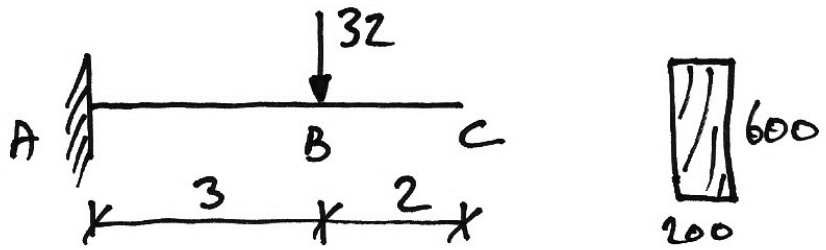
Again, sketch this:

3. Application to Determinate Structures

3.1 Basic Examples

Example 3

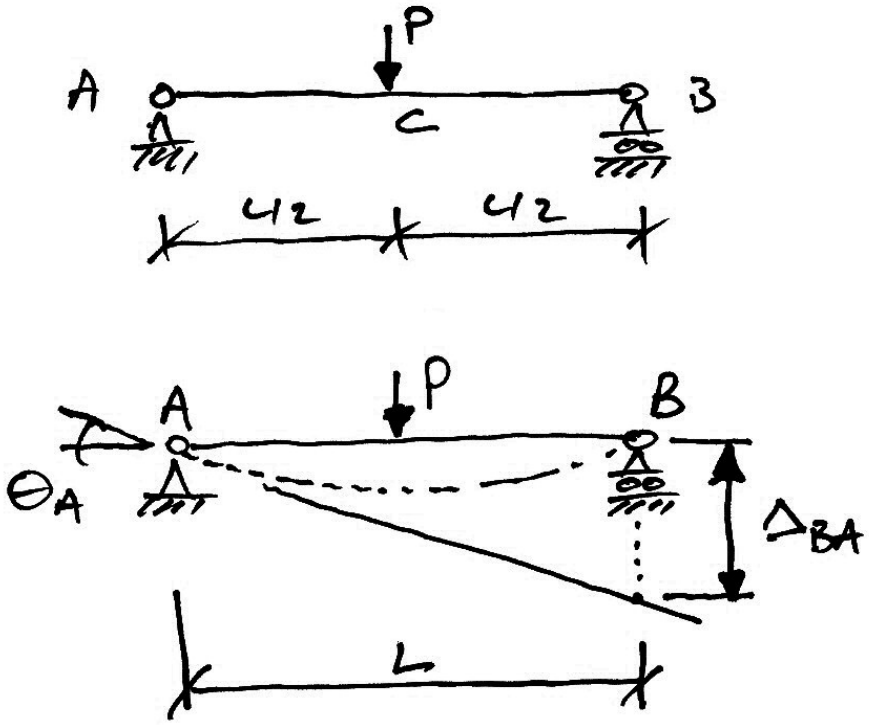
For the following beam, find δ_B, δ_C, θ_B and θ_C given the section dimensions shown and $E = 10$ kN/mm^2.



To be done in class.

Example 4

For the following simply-supported beam, we can find the rotation at *A* using Mohr's Second Theorem. The deflected shape diagram is used to identify relationships between vertical intercepts and rotations:

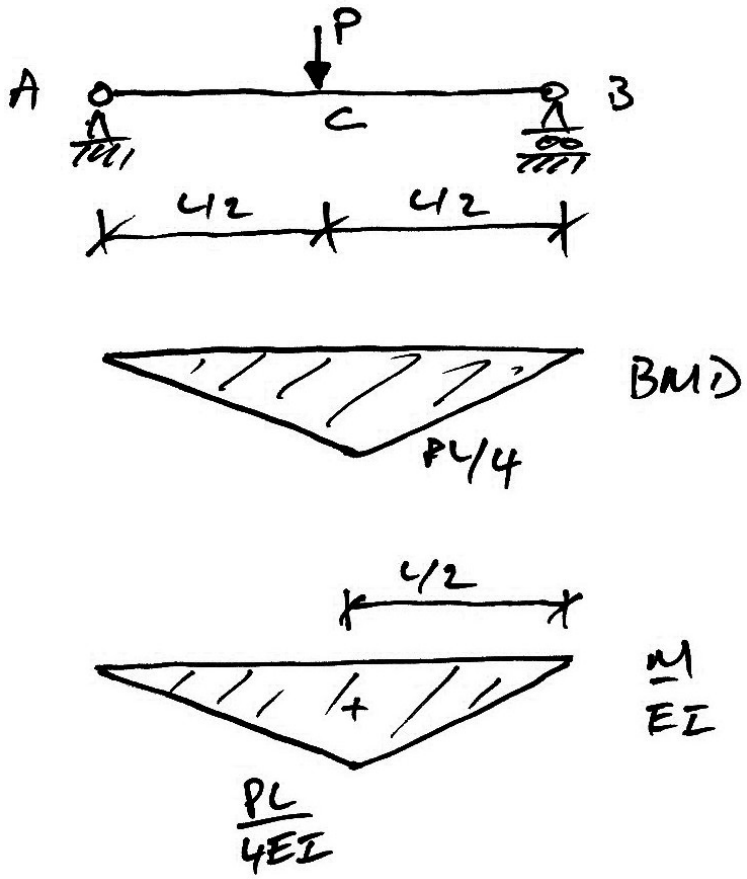


The key to the solution here is that we can calculate Δ_{BA} using Mohr II but from the diagram we can see that we can use the formula $S = R\theta$ for small angles:

$$\Delta_{BA} = L \cdot \theta_A$$

Therefore once we know Δ_{BA} using Mohr II, we can find $\theta_A = \Delta_{BA}/L$.

To calculate Δ_{BA} using Mohr II we need the bending moment and curvature diagrams:



Thus, from Mohr II, we have:

$$\Delta_{BA} = \left[\frac{1}{2}\cdot L\cdot\frac{PL}{4EI}\right]\left[\frac{L}{2}\right]$$
$$= \frac{PL^3}{16EI}$$

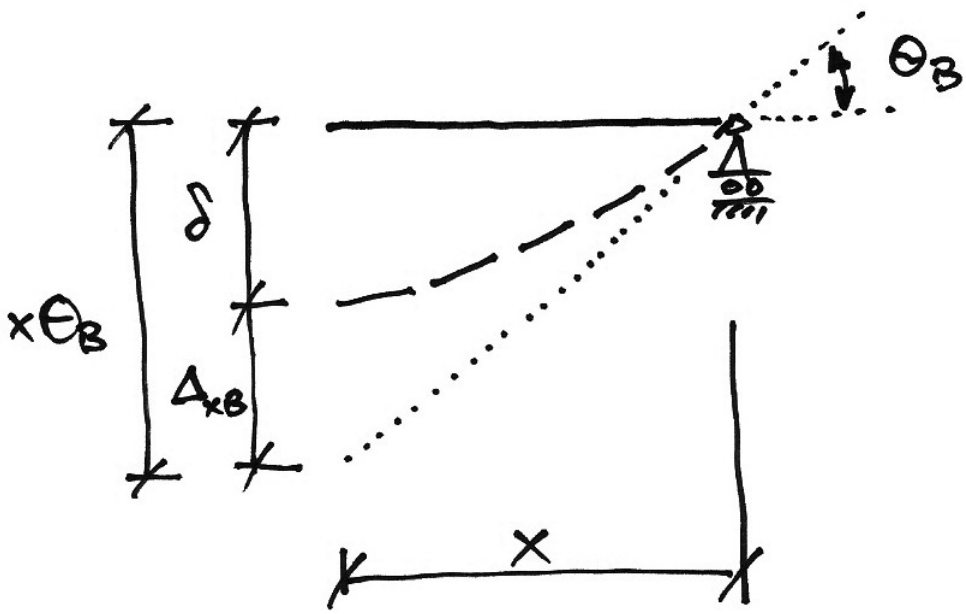
But, $\Delta_{BA} = L\cdot\theta_A$ and so we have:

$$\theta_A = \frac{\Delta_{BA}}{L}$$
$$= \frac{PL^2}{16EI}$$

3.2 Finding Deflections

General Procedure

To find the deflection at any location *x* from a support use the following relationships between rotations and vertical intercepts:



Thus we:

1. Find the rotation at the support using Mohr II as before;
2. For the location *x*, and from the diagram we have:

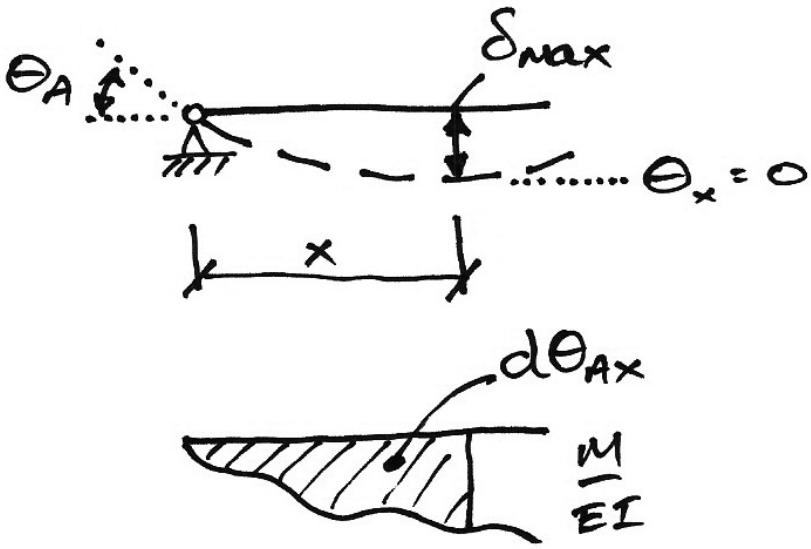
$$\delta_x = x \cdot \theta_B - \Delta_{xB}$$

Maximum Deflection

To find the maximum deflection we first need to find the location at which this occurs. We know from beam theory that:

$$\delta = \frac{d\theta}{dx}$$

Hence, from basic calculus, the maximum deflection occurs at a rotation, $\theta = 0$:



To find where the rotation is zero:

1. Calculate a rotation at some point, say support *A*, using Mohr II say;
2. Using Mohr I, determine at what distance from the point of known rotation (*A*) the change in rotation (Mohr I), $d\theta_{Ax}$ equals the known rotation (θ_A).
3. This is the point of maximum deflection since:

$$\theta_A - d\theta_{Ax} = \theta_A - \theta_A = 0$$

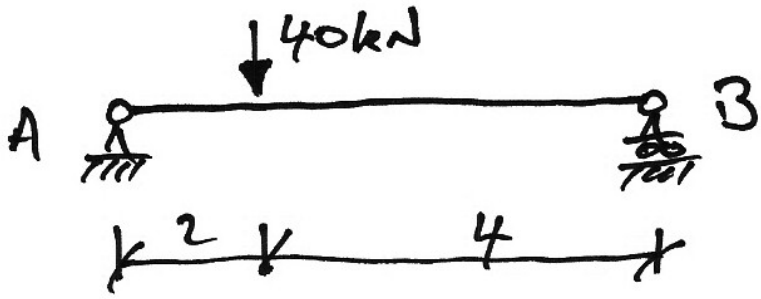
Example 5

For the following beam of constant *EI*:

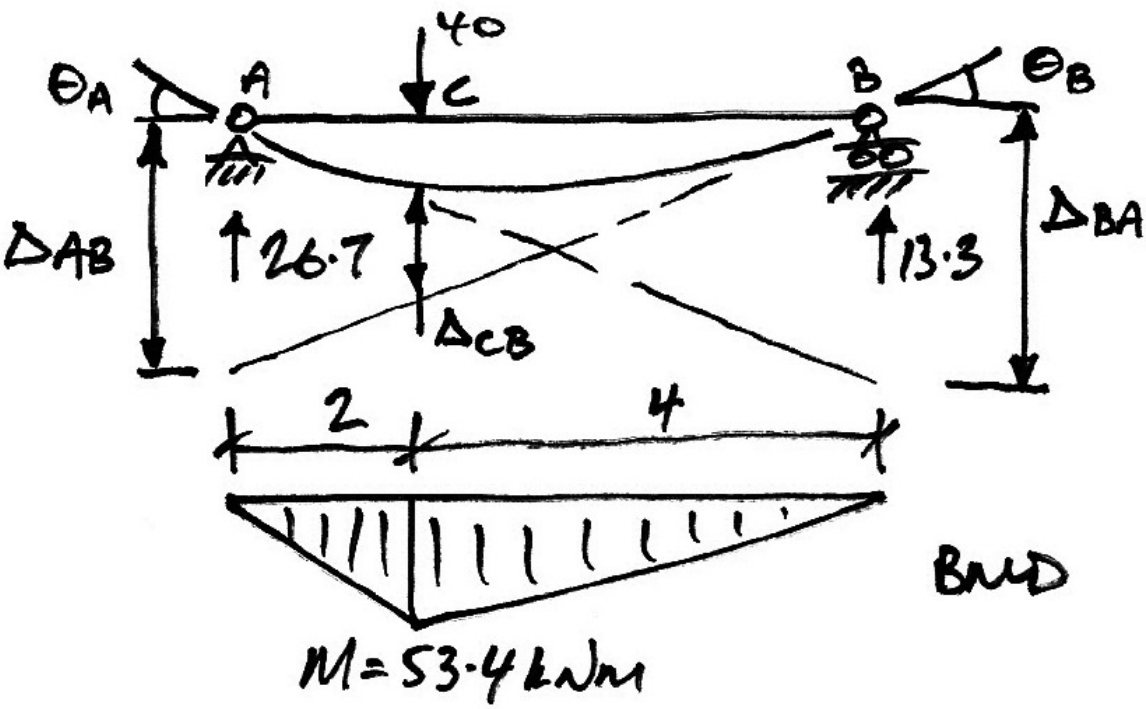
(a) Determine θ_A, θ_B and δ_C;

(b) What is the maximum deflection and where is it located?

Give your answers in terms of *EI*.



The first step is to determine the BMD and draw the deflected shape diagram with rotations and tangents indicated:



Rotations at *A* and *B*

To calculate the rotations, we need to calculate the vertical intercepts and use the fact that the intercept is length times rotation. Thus, for the rotation at *B*:

$$
\begin{aligned}
EI\Delta_{AB} &= \left(\frac{2}{3}\cdot 2\right)\left(\frac{1}{2}\cdot 2\cdot M\right)+\left(2+\frac{4}{3}\right)\left(\frac{1}{2}\cdot 4\cdot M\right) \\
&= M\left(\frac{4}{3}+\frac{20}{3}\right) \\
&= 8M \\
\therefore \Delta_{AB} &= \frac{8M}{EI}
\end{aligned}
$$

But, we also know that $\Delta_{AB} = 6\theta_B$. Hence:

$$
\begin{aligned}
6\theta_B &= \frac{8M}{EI} \\
\therefore \theta_B &= \frac{4M}{3EI} = 1.33\frac{M}{EI}
\end{aligned}
$$

Similarly for the rotation at *A*:

$$
\begin{aligned}
EI\Delta_{BA} &= \left(\frac{2}{3}\cdot 4\right)\left(\frac{1}{2}\cdot 4\cdot M\right)+\left(4+\frac{1}{3}\cdot 2\right)\left(\frac{1}{2}\cdot 2\cdot M\right) \\
&= M\left(\frac{16}{3}+\frac{14}{3}\right) \\
&= 10M \\
\therefore \Delta_{BA} &= \frac{10M}{EI}
\end{aligned}
$$

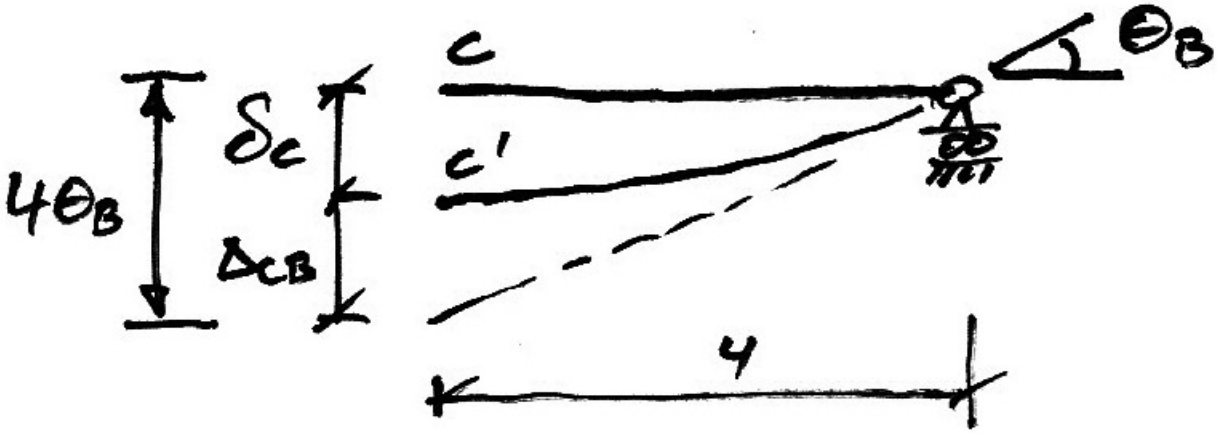
But, we also know that $\Delta_{BA} = 6\theta_A$ and so:

$$6\theta_A = \frac{10M}{EI}$$

$$\therefore \theta_A = \frac{5M}{3EI} = 1.67\frac{M}{EI}$$

Deflection at *C*

To find the deflection at *C*, we use the vertical intercept Δ_{CB} and θ_B:



From the figure, we see:

$$\delta_C = 4\theta_B - \Delta_{CB}$$

And so from the BMD and rotation at *B*:

$$EI\delta_C = 4(1.33M) - \left(\frac{1}{2}\cdot 4\cdot M\right)\left(\frac{4}{3}\right)$$

$$\therefore \delta_C = 2.665\frac{M}{EI}$$

Maximum Deflection

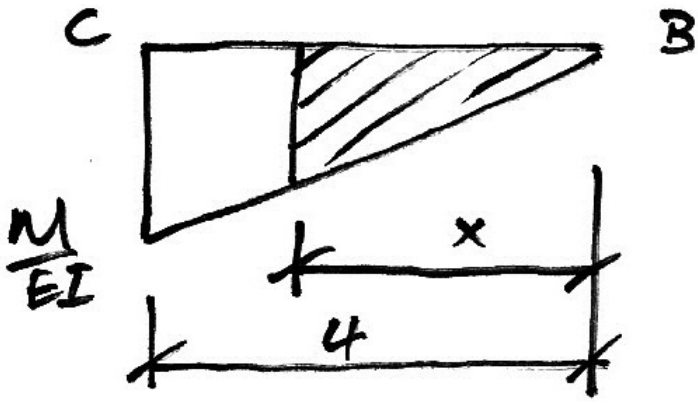
The first step in finding the maximum deflection is to locate it. We know tow things:

1. Maximum deflection occurs where there is zero rotation;
2. Maximum deflection is always close to the centre of the span.

Based on these facts, we work with Mohr I to find the point of zero rotation, which will be located between *B* and *C*, as follows:

$$\text{Change in rotation} = \theta_B - 0 = \theta_B$$

But since we know that the change in rotation is also the area of the M/EI diagram we need to find the point *x* where the area of the M/EI diagram is equal to θ_B:



Thus:

$$EI\left(\theta_B - 0\right) = \left(M \cdot \frac{x}{4}\right) \cdot \frac{1}{2} \cdot x$$

$$EI\theta_B = M\frac{x^2}{8}$$

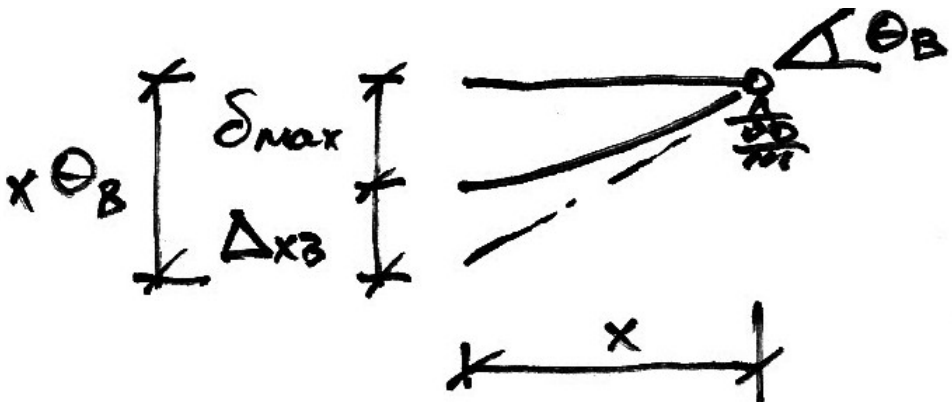
But we know that $\theta_B = 1.33\dfrac{M}{EI}$, hence:

$$EI\left(1.33\frac{M}{EI}\right)=M\,\frac{x^2}{8}$$

$$x^2=10.66$$

$$x=3.265 \text{ m from } B \text{ or } 2.735 \text{ m from } A$$

So we can see that the maximum deflection is 265 mm shifted from the centre of the beam towards the load. Once we know where the maximum deflection is, we can calculate is based on the following diagram:



Thus:

$$\delta_{\max}=x\theta_B-\Delta_{xB}$$

$$EI\delta_{\max}=x\left(1.33M\right)-\left(M\,\frac{x^2}{8}\right)\left(\frac{x}{3}\right)$$

$$=M\left(4.342-1.450\right)$$

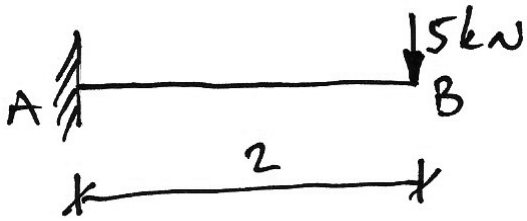
$$\delta_{\max}=2.892\frac{M}{EI}$$

And since $M=53.4$ kNm, $\delta_{\max}=\dfrac{154.4}{EI}$.

3.3 Problems

1. For the beam of Example 3, using only Mohr's First Theorem, show that the rotation at support *B* is equal in magnitude but not direction to that at *A*.

2. For the following beam, of dimensions $b = 150$ mm and $d = 225$ mm and $E = 10$ kN/mm^2, show that $\theta_B = 7 \times 10^{-4}$ rads and $\delta_B = 9.36$ mm.



3. For a cantilever *AB* of length *L* and stiffness *EI*, subjected to a UDL, show that:

$$\theta_B = \frac{wL^3}{6EI}; \quad \delta_B = \frac{wL^4}{8EI}$$

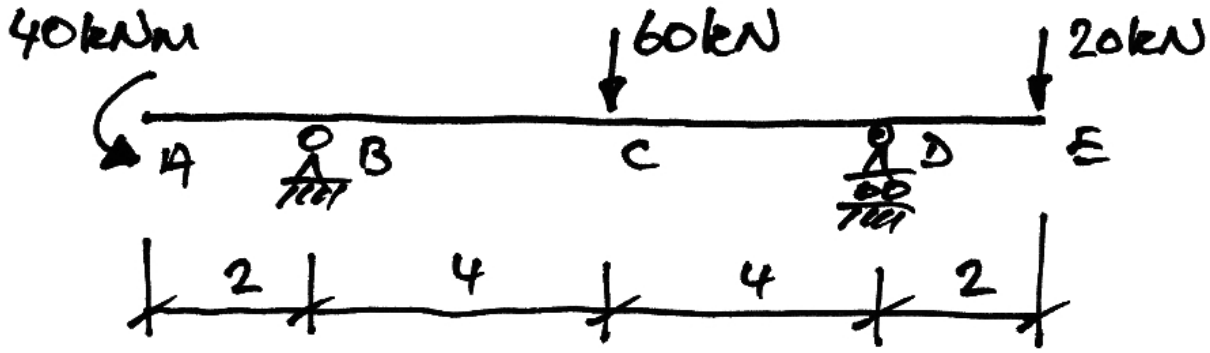
4. For a simply-supported beam *AB* with a point load at mid span (*C*), show that:

$$\delta_C = \frac{PL^3}{48EI}$$

5. For a simply-supported beam *AB* of length *L* and stiffness *EI*, subjected to a UDL, show that:

$$\theta_A = \frac{wL^3}{24EI}; \quad \theta_B = -\frac{wL^3}{24EI}; \quad \delta_C = \frac{5wL^4}{384EI}$$

6. For the following beam, determine the deflections at *A*, *E* and the maximum overall deflection in the span. Take $EI = 40\ \text{MNm}^2$



Ans. 6.00 mm, 2.67 mm, 8.00 mm

4. Application to Indeterminate Structures

4.1 Basis of Approach

Using the principle of superposition we will separate indeterminate structures into a primary and reactant structures.

For these structures we will calculate the deflections at a point for which the deflection is known in the original structure.

We will then use compatibility of displacement to equate the two calculated deflections to the known deflection in the original structure.

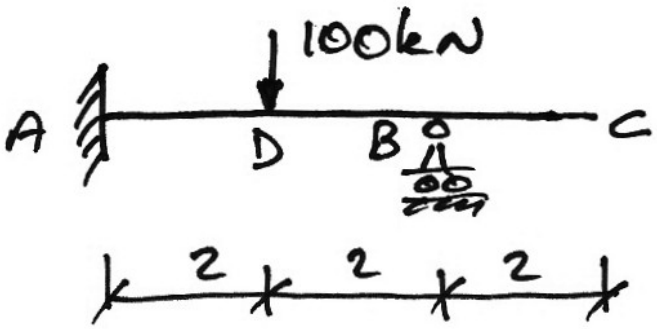
Doing so will yield the value of the redundant reaction chosen for the reactant structure.

Once this is known all other load effects (bending, shear, deflections, rotations) can be calculated.

See the notes on the Basis for the Analysis of Indeterminate Structures for more on this approach.

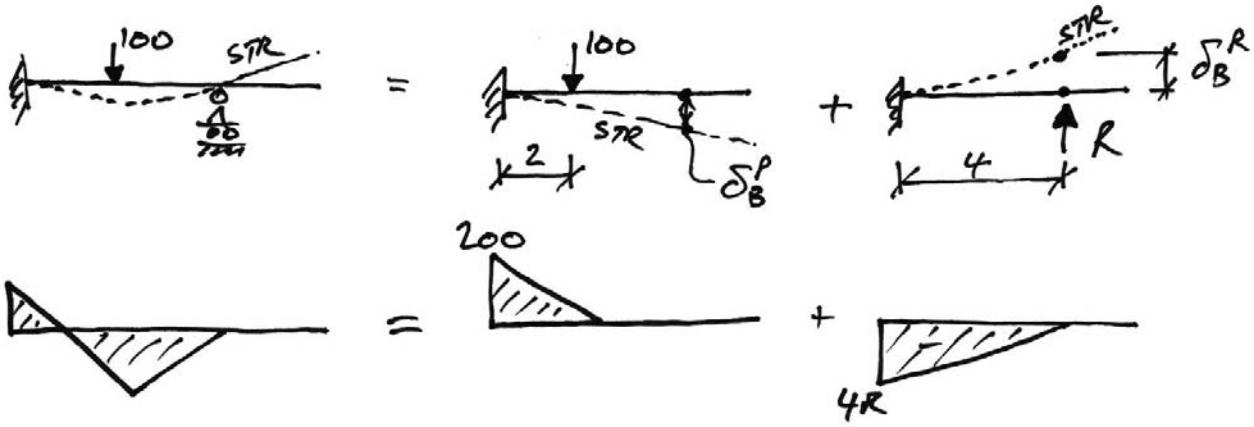
4.2 Example 6: Propped Cantilever

For the following prismatic beam, find the maximum deflection in span *AB* and the deflection at C in terms of *EI*.



Find the reaction at B

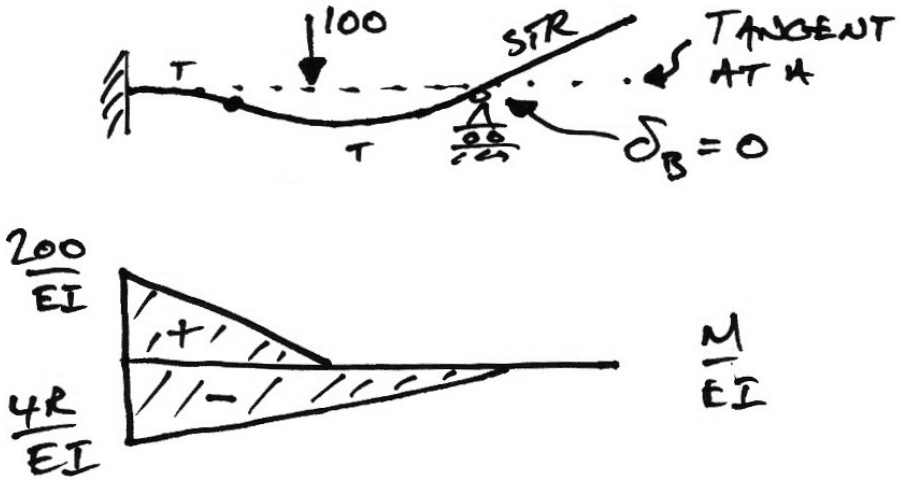
Since this is an indeterminate structure, we first need to solve for one of the unknown reactions. Choosing V_B as our redundant reaction, using the principle of superposition, we can split the structure up as shown:



Final = Primary + Reactant

In which *R* is the value of the chosen redundant.

In the final structure (a) we know that the deflection at *B*, δ_B, must be zero as it is a roller support. So from the BMD that results from the superposition of structures (b) and (c) we can calculate δ_B in terms of *R* and solve since $\delta_B = 0$.



We have from Mohr II:

$$
\begin{aligned}
EI\Delta_{BA} &= \left[\left(\frac{1}{2}\cdot 2\cdot 200\right)\left(2+\frac{2}{3}\cdot 2\right)\right]_{(b)} + \left[-\left(\frac{1}{2}\cdot 4\cdot 4R\right)\left(\frac{2}{3}\cdot 4\right)\right]_{(c)} \\
&= \frac{2000}{3} - \frac{64}{3}R \\
&= \frac{1}{3}\left(2000 - 64R\right)
\end{aligned}
$$

But since $\theta_A = 0$, $\delta_B = \Delta_{BA} = 0$ and so we have:

$$
\begin{aligned}
EI\Delta_{BA} &= 0 \\
\frac{1}{3}\left(2000 - 64R\right) &= 0 \\
64R &= 2000 \\
R &= +31.25 \text{ kN}
\end{aligned}
$$

The positive sign for R means that the direction we originally assumed for it (upwards) was correct.

At this point the final BMD can be drawn but since its shape would be more complex we continue to operate using the structure (b) and (c) BMDs.

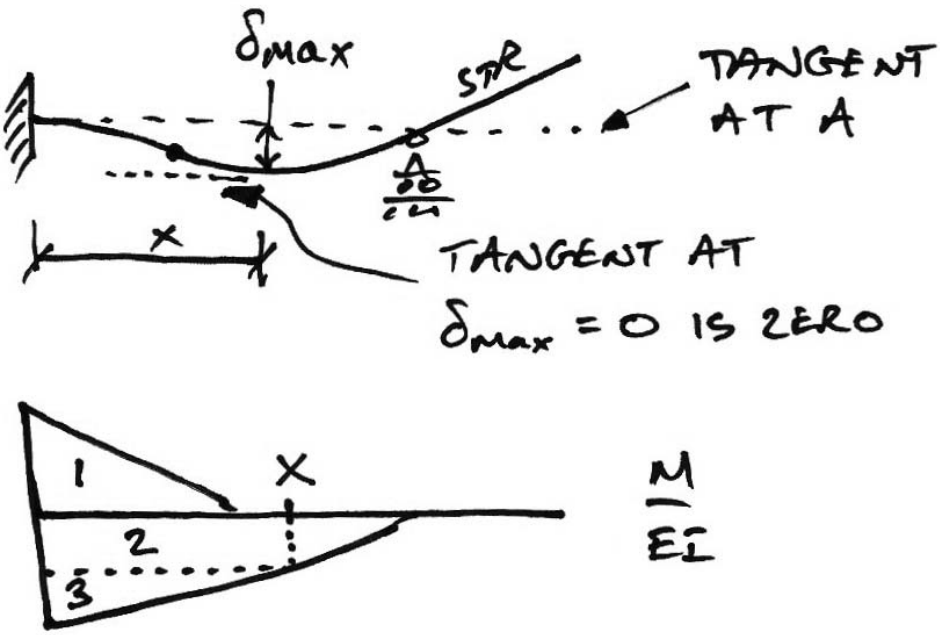
Draw the Final BMD

Find the location of the maximum deflection

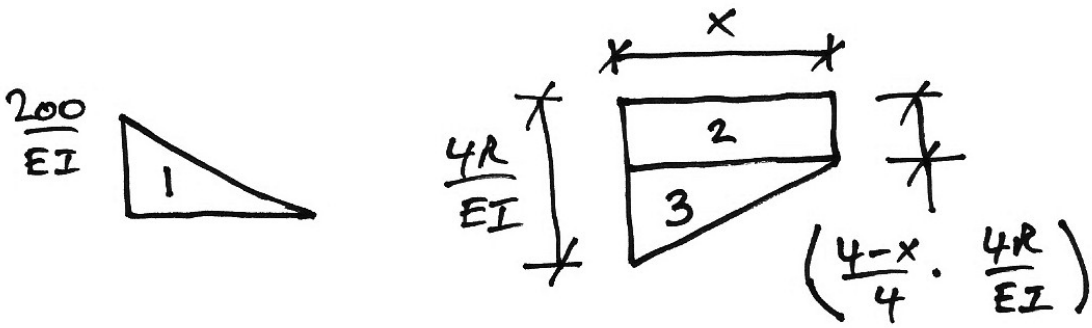
This is the next step in determining the maximum deflection in span *AB*. Using the knowledge that the tangent at *A* is horizontal, i.e. $\theta_A = 0$, we look for the distance *x* from *A* that satisfies:

$$d\theta_{Ax} = \theta_A - \theta_x = 0$$

By inspection on the deflected shape, it is apparent that the maximum deflection occurs to the right of the point load. Hence we have the following:



So using Mohr I we calculate the change in rotation by finding the area of the curvature diagram between *A* and *x*. The diagram is split for ease:



The Area 1 is trivial:

$$A_1 = \frac{1}{2} \cdot 2 \cdot \frac{200}{EI} = \frac{200}{EI}$$

For Area 2, we need the height first which is:

$$h_2 = \frac{4-x}{4} \cdot \frac{4R}{EI} = \frac{4 \cdot 125 - 125}{4EI} = \frac{125}{EI} - \frac{125}{EI}x$$

And so the area itself is:

$$A_2 = x \cdot \left[\frac{125}{EI} - \frac{125}{EI}x \right]$$

For Area 3 the height is:

$$h_3 = \frac{125}{EI} - \left[\frac{125}{EI} - \frac{125}{EI}x \right] = \frac{125}{EI}x$$

And so the area is:

$$A_2 = \frac{1}{2} \cdot x \cdot \frac{125}{EI} x$$

Being careful of the signs for the curvatures, the total area is:

$$\begin{aligned} EId\theta_{Ax} &= -A_1 + A_2 + A_3 \\ &= -200 + x\left(125 - \frac{125}{4}x\right) + \frac{125}{8}x^2 \\ &= \left(\frac{125}{8} - \frac{125}{4}\right)x^2 + 125x - 200 \end{aligned}$$

Setting this equal to zero to find the location of the maximum deflection, we have:

$$\begin{aligned} -\frac{125}{8}x^2 + 125x - 200 &= 0 \\ 5x^2 - 40x + 64 &= 0 \end{aligned}$$

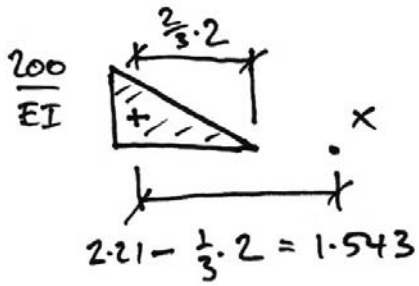
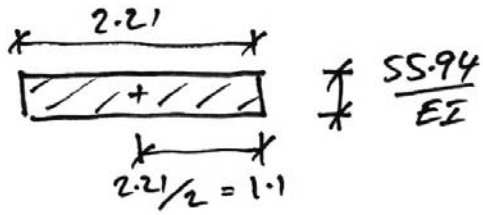
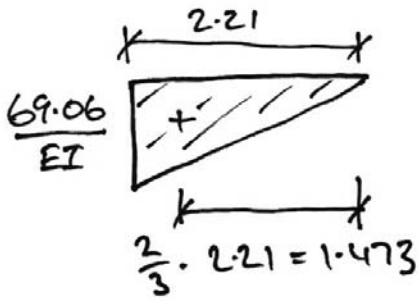
Thus, $x = 5.89\text{ m}$ or $x = 2.21\text{ m}$. Since we are dealing with the portion *AB*, $x = 2.21\text{ m}$.

Find the maximum deflection

Since the tangent at both *A* and *x* are horizontal, i.e. $\theta_A = 0$ and $\theta_x = 0$, the deflection is given by:

$$\delta_{\max} = \Delta_{xA}$$

Using Mohr II and Areas 1, 2 and 3 as previous, we have:

Area 1		$A_1\bar{x}_1 = -\frac{200}{EI}\cdot 1.543$ $= -\frac{308.67}{EI}$
Area 2		$h_2 = \frac{4-2.21}{4}\cdot\frac{4R}{EI} = \frac{55.94}{EI}$ $A_2\bar{x}_2 = 2.21\cdot\frac{55.94}{EI}\cdot\frac{2.21}{2}$ $= \frac{136.61}{EI}$
Area 3		$h_3 = 2.21\cdot\frac{125}{EI} = \frac{69.06}{EI}$ $A_3\bar{x}_3 = \left[\frac{1}{2}\cdot 2.21\cdot\frac{69.06}{EI}\right]\cdot 1.473$ $= \frac{112.43}{EI}$

Thus:

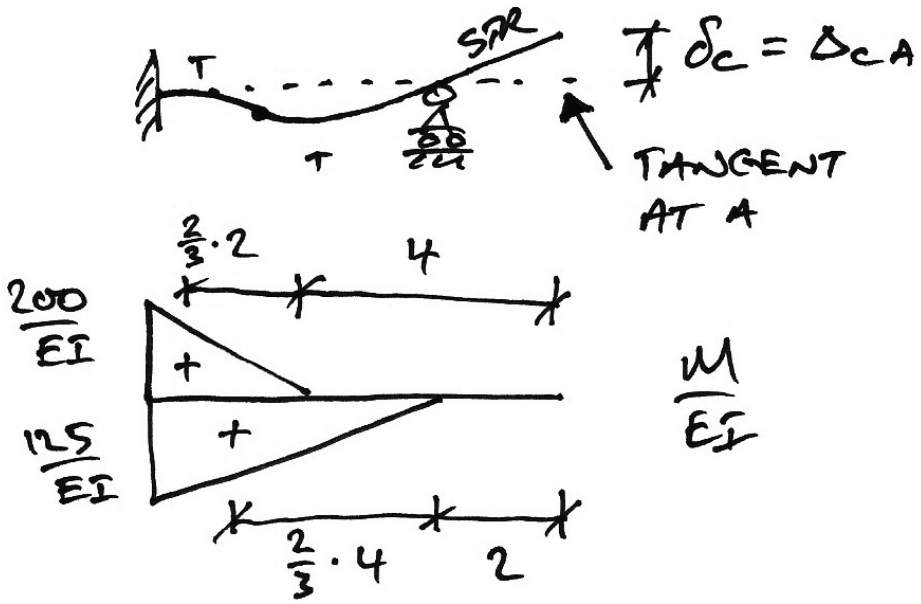
$$EI\Delta_{xB} = EI\delta_{\max} = -308.67 + 136.61 + 112.43$$
$$\Rightarrow \delta_{\max} = \frac{-59.63}{EI}$$

The negative sign indicates that the negative bending moment diagram dominates, i.e. the hogging of the cantilever is pushing the deflection downwards.

Find the deflection at C

For the deflection at C we again use the fact that $\theta_A = 0$ with Mohr II to give:

$$\delta_C = \Delta_{CA}$$



From the diagram we have:

$$EI\Delta_{CA} = -\left(\frac{1}{2}\cdot 2\cdot 200\right)\left(\frac{4}{3}+4\right)+\left(\frac{1}{2}\cdot 4\cdot 125\right)\left(2+\frac{8}{3}\right)$$

$$\delta_C = \frac{+100}{EI}$$

The positive sign indicates that the positive bending moment region dominates and so the deflection is upwards.

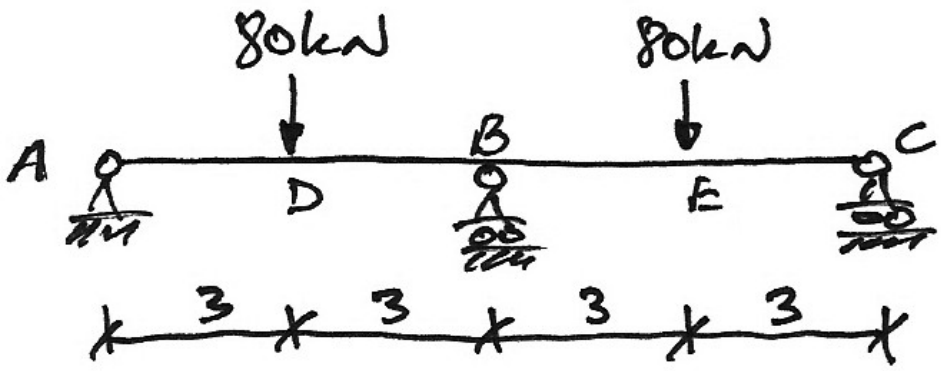
4.3 Example 7: 2-Span Beam

For the following beam of constant *EI*, using Mohr's theorems:

(a) Draw the bending moment diagram;

(b) Determine, δ_D and δ_E;

Give your answers in terms of *EI*.

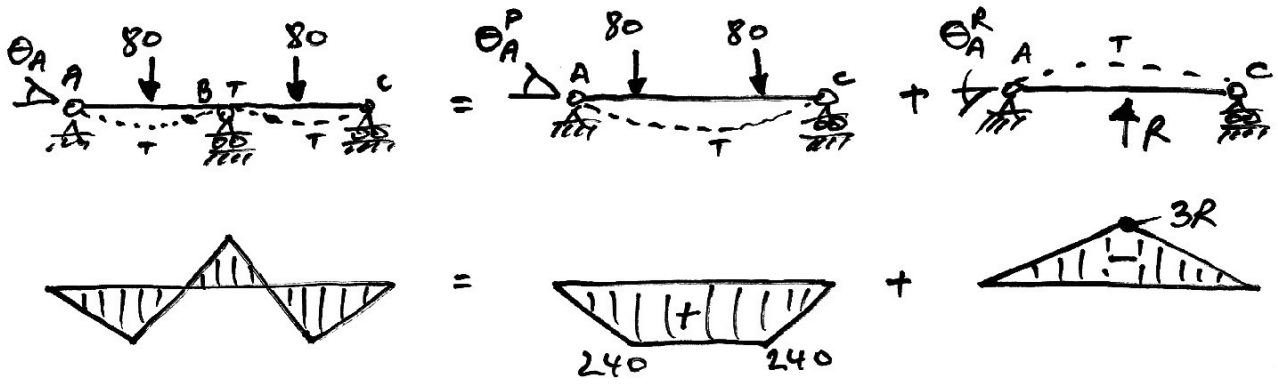


In the last example we knew the rotation at *A* and this made finding the deflection at the redundant support relatively easy. Once again we will choose a redundant support, in this case the support at *B*.

In the present example, we do not know the rotation at *A* – it must be calculated – and so finding the deflection at *B* is more involved. We can certainly use compatibility of displacement at *B*, but in doing so we will have to calculate the vertical intercept from *B* to *A*, Δ_{BA}, twice. Therefore, to save effort, we use Δ_{BA} as the measure which we apply compatibility of displacement to. We will calculate Δ_{BA} through calculation of θ_A (and using the small angle approximation) and through direct calculation from the bending moment diagram. We will then have an equation in *R* which can be solved.

Rotation at *A*

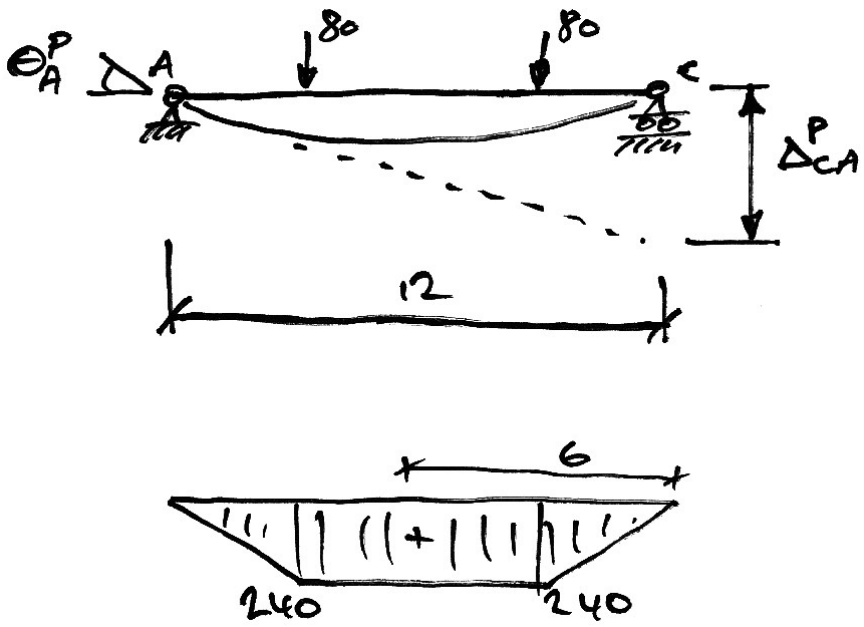
Breaking the structure up into primary and redundant structures:



So we can see that the final rotation at *A* is:

$$\theta_A = \theta_A^P + \theta_A^R$$

To find the rotation at *A* in the primary structure, consider the following:



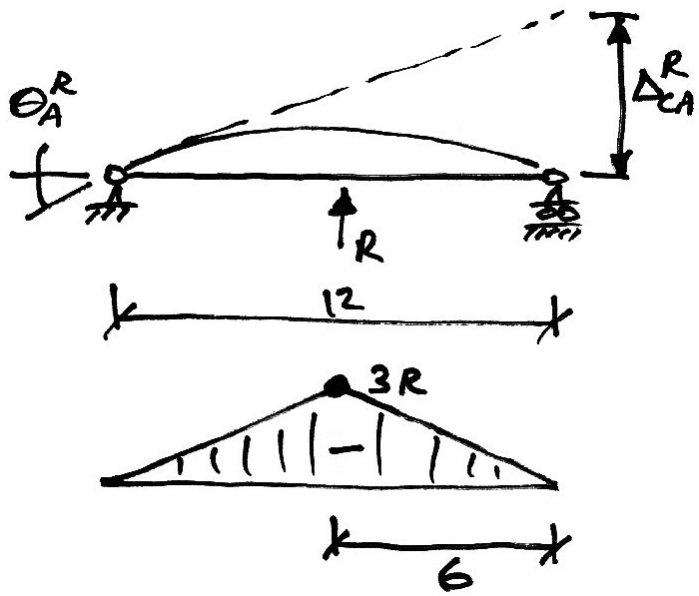
By Mohr II we have:

$$EI\Delta_{CA} = (240 \cdot 9)(6) = 12960$$

But we know, from the small angle approximation, $\Delta_{CA} = 12\theta_A$, hence:

$$EI\theta_A^P = \frac{\Delta_{CA}}{12} = \frac{12960}{12} = 1080$$
$$\therefore \theta_A^P = \frac{1080}{EI}$$

To find the rotation at *A* for the reactant structure, we have:



$$EI\Delta_{CA} = \left(\frac{1}{2} \cdot 12 \cdot 3R\right)(6) = 108R$$

$$\Delta_{CA} = 12\theta_A$$

$$EI\theta_A^R = \frac{\Delta_{CA}}{12} = \frac{108R}{12} = 9R$$
$$\therefore \theta_A^R = -\frac{9R}{EI}$$

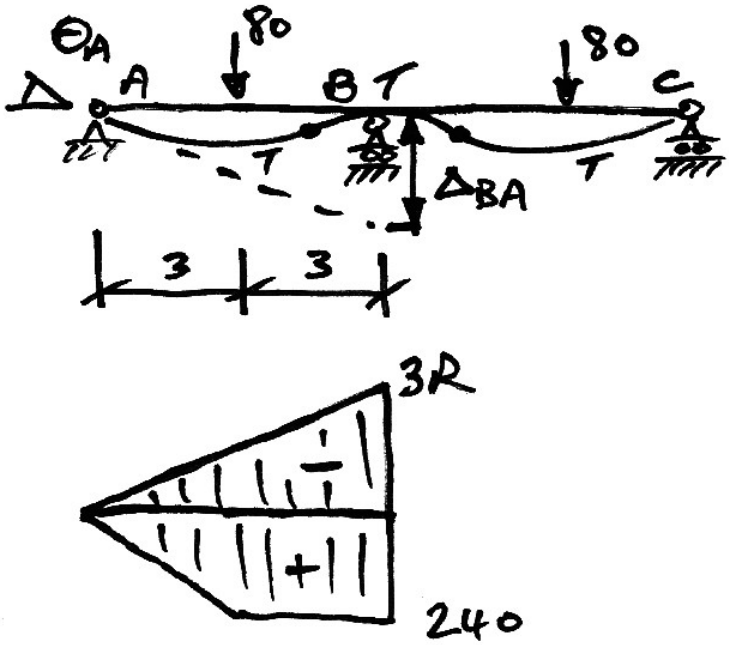
Notice that we assign a negative sign to the reactant rotation at *A* since it is in the opposite sense to the primary rotation (which we expect to dominate).

Thus, we have:

$$\begin{aligned}\theta_A &= \theta_A^P + \theta_A^R \\ &= \frac{1080}{EI} - \frac{9R}{EI}\end{aligned}$$

Vertical Intercept from *B* to *A*

The second part of the calculation is to find Δ_{BA} directly from calculation of the curvature diagram:



Thus we have:

$$EI\Delta_{BA} = -\left(\frac{1}{2}\cdot 6\cdot 3R\right)\left(\frac{1}{3}\cdot 6\right) + \left(240\cdot 3\right)\left(\frac{3}{2}\right) + \left(\frac{1}{2}\cdot 3\cdot 240\right)\left(3+\frac{3}{3}\right)$$

$$EI\Delta_{BA} = -18R + 1080 + 1440$$
$$\therefore \Delta_{BA} = \frac{2520 - 18R}{EI}$$

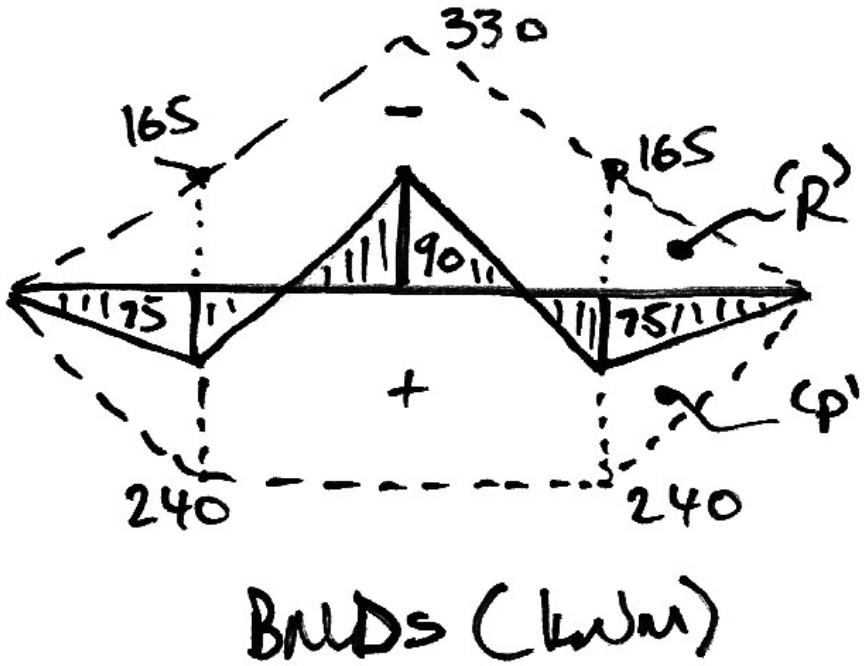
Solution for *R*

Now we recognize that $\Delta_{BA} = 6\theta_A$ by compatibility of displacement, and so:

$$\frac{2520 - 18R}{EI} = 6\left(\frac{1080}{EI} - \frac{9R}{EI}\right)$$
$$2520 - 18R = 6(1080 - 9R)$$
$$36R = 3960$$
$$R = 110 \text{ kN}$$

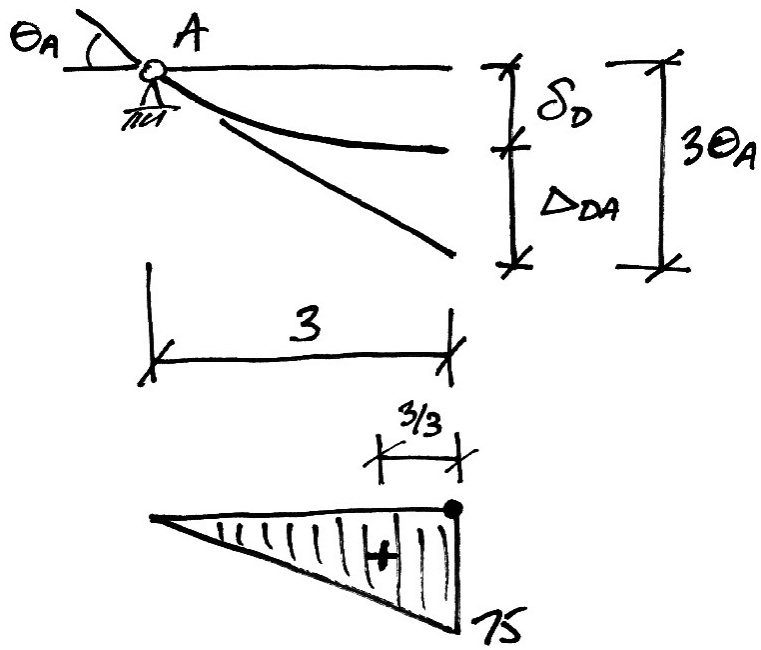
Solution to Part (a)

With this we can immediately solve for the final bending moment diagram by superposition of the primary and reactant BMDs:



Solution to Part (b)

We are asked to calculate the deflection at *D* and *E*. However, since the beam is symmetrical $\delta_D = \delta_E$ and so we need only calculate one of them – say δ_D. Using the (now standard) diagram for the calculation of deflection:



$$\theta_A = \frac{1080}{EI} - \frac{9(110)}{EI} = \frac{90}{EI}$$

$$EI\Delta_{DA} = \left(\frac{1}{3}\cdot 3\cdot 75\right)\left(\frac{3}{3}\right) = 112.5$$

But $\delta_D = 3\theta_A - \Delta_{DA}$, thus:

$$\begin{aligned} EI\delta_D &= 3(90) - 112.5 \\ &= 157.5 \end{aligned}$$

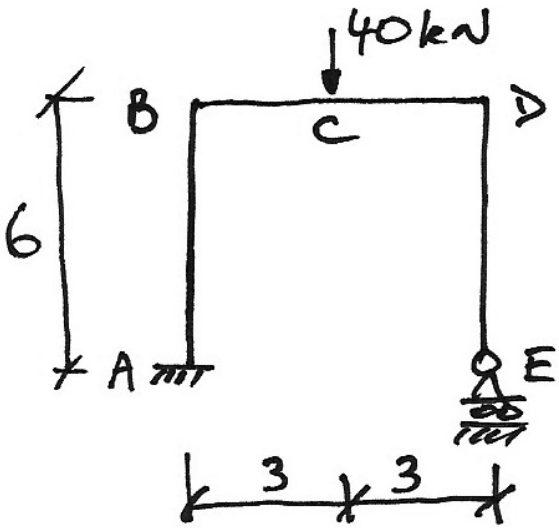
$$\delta_D = \delta_E = \frac{157.5}{EI}$$

4.4 Example 8: Simple Frame

For the following frame of constant $EI = 40\ \text{MNm}^2$, using Mohr's theorems:

(a) Draw the bending moment and shear force diagram;

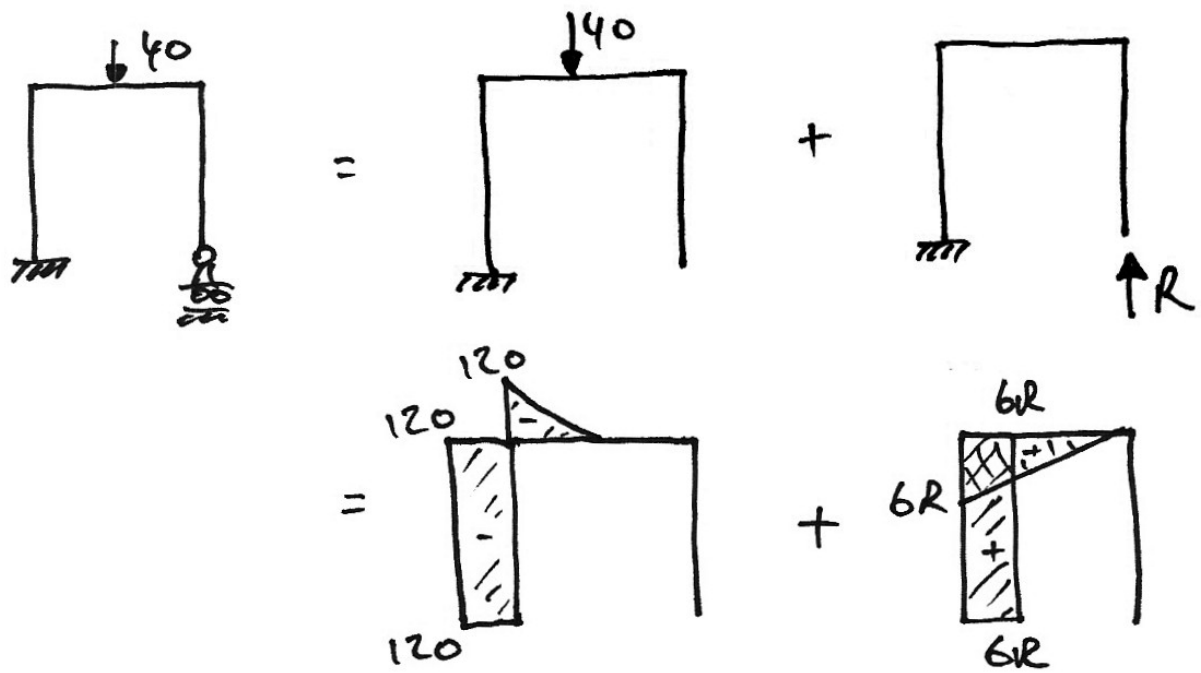
(b) Determine the horizontal deflection at *E*.



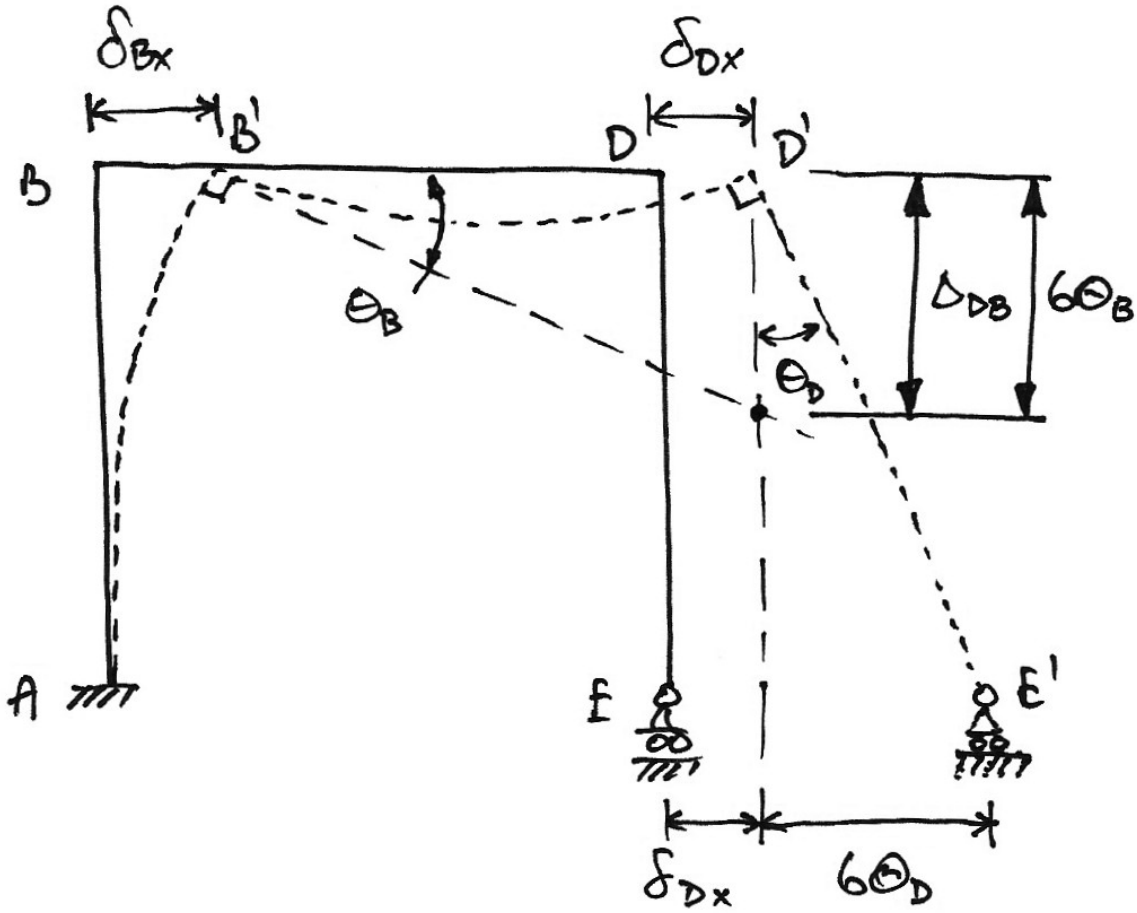
Part (a)

Solve for a Redundant

As with the beams, we split the structure into primary and reactant structures:



We also need to draw the deflected shape diagram of the original structure to identify displacements that we can use:

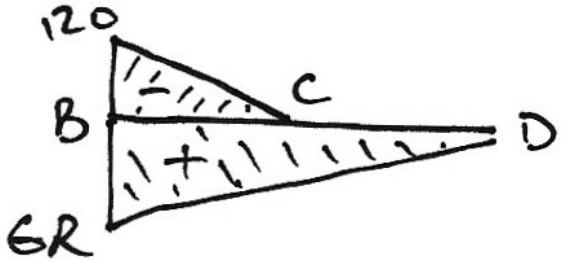


To solve for *R* we could use any known displacement. In this case we will use the vertical intercept Δ_{DB} as shown, because:

- We can determine Δ_{DB} for the original structure in terms of *R* using Mohr's Second Theorem;
- We see that $\Delta_{DB} = 6\theta_B$ and so using Mohr's First Theorem for the original structure we will find θ_B, again in terms of *R*;
- We equate the two methods of calculating Δ_{DB} (both are in terms of *R*) and solve for *R*.

Find Δ_{DB} by Mohr II

Looking at the combined bending moment diagram, we have:



$$
\begin{aligned}
EI\Delta_{DB} &= \left[\frac{1}{2}\cdot 6\cdot 6R\right]\cdot\left[\frac{2}{3}\cdot 6\right]-\left[\frac{1}{2}\cdot 3\cdot 120\right]\cdot\left[3+\frac{2}{3}\cdot 3\right] \\
&= 72R-900
\end{aligned}
$$

Find θ_B by Mohr I

Since the tangent at *A* is vertical, the rotation at *B* will be the change in rotation from *A* to *B*:

$$
\begin{aligned}
d\theta_{BA} &= \theta_B-\theta_A \\
&= \theta_B-0 \\
&= \theta_B \\
&= \text{Area of}\left(\frac{M}{EI}\right)_{B\text{ to }A}
\end{aligned}
$$

Therefore, by Mohr I:

$$
\begin{aligned}
EI\theta_B &= \text{Area of}\left(\frac{M}{EI}\right)_{B\text{ to }A} \\
&= 6\cdot 6R-120\cdot 6 \\
&= 36R-720
\end{aligned}
$$

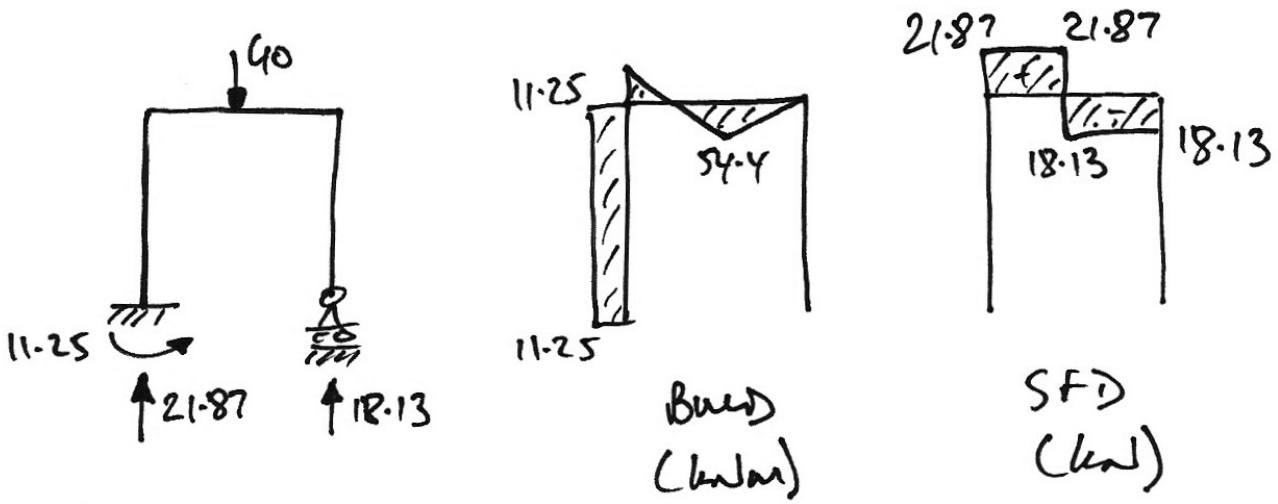
Equate and Solve for *R*

As identified previously:

$$\Delta_{DB} = 6\theta_B$$
$$72R - 900 = 6\left[36R - 720\right]$$
$$R = 18.13 \text{ kN}$$

Diagrams

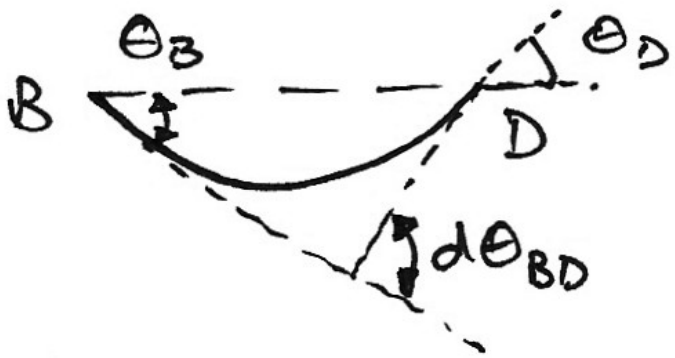
Knowing *R* we can then solve for the reactions, bending moment and shear force diagrams. The results are:



Part (b)

The movement at *E* is comprised of δ_{Dx} and $6\theta_D$ as shown in the deflection diagram. These are found as:

- Since the length of member *BD* doesn't change, $\delta_{Dx} = \delta_{Bx}$. Further, by Mohr II, $\delta_{Bx} = \Delta_{BA}$;
- By Mohr I, $\theta_D = \theta_B - d\theta_{BD}$, that is, the rotation at *D* is the rotation at *B* minus the change in rotation from *B* to *D*:



So we have:

$$\begin{aligned} EI\Delta_{BA} &= [6R \cdot 6][3] - [120 \cdot 6][3] \\ &= -202.5 \end{aligned}$$

$$\begin{aligned} EId\theta_{BD} &= \left[\frac{1}{2} \cdot 6R \cdot 6\right] - \left[\frac{1}{2} \cdot 120 \cdot 3\right] \\ &= 146.25 \end{aligned}$$

Notice that we still use the primary and reactant diagrams even though we know *R*. We do this because the shapes and distances are simpler to deal with.

From before we know:

$$EI\theta_B = 36R - 720 = 67.5$$

Thus, we have:

$$\begin{aligned} EI\theta_D &= EI\theta_B - d\theta_{BD} \\ &= 67.5 - 146.25 \\ &= -78.75 \end{aligned}$$

The minus indicates that it is a rotation in opposite direction to that of θ_B which is clear from the previous diagram. Since we have taken account of the sense of the rotation, we are only interested in its absolute value. A similar argument applies to the minus sign for the deflection at *B*. Therefore:

$$\begin{aligned} \delta_{Ex} &= \delta_{Bx} + 6\theta_D \\ &= \frac{202.5}{EI} + 6 \cdot \frac{78.75}{EI} \\ &= \frac{675}{EI} \end{aligned}$$

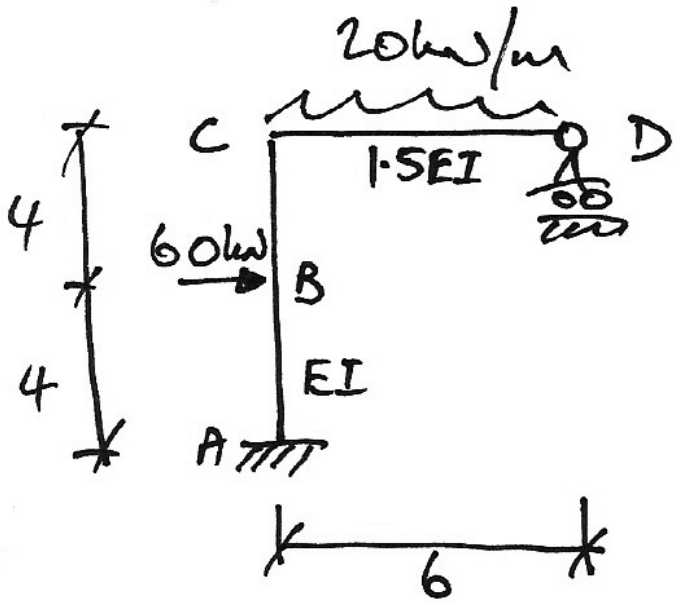
Using $EI = 40 \text{ MNm}^2$ gives $\delta_{Ex} = 16.9 \text{ mm}$.

4.5 Example 9: Complex Frame

For the following frame of constant $EI = 40\ \text{MNm}^2$, using Mohr's theorems:

(a) Determine the reactions and draw the bending moment diagram;

(b) Determine the horizontal deflection at *D*.

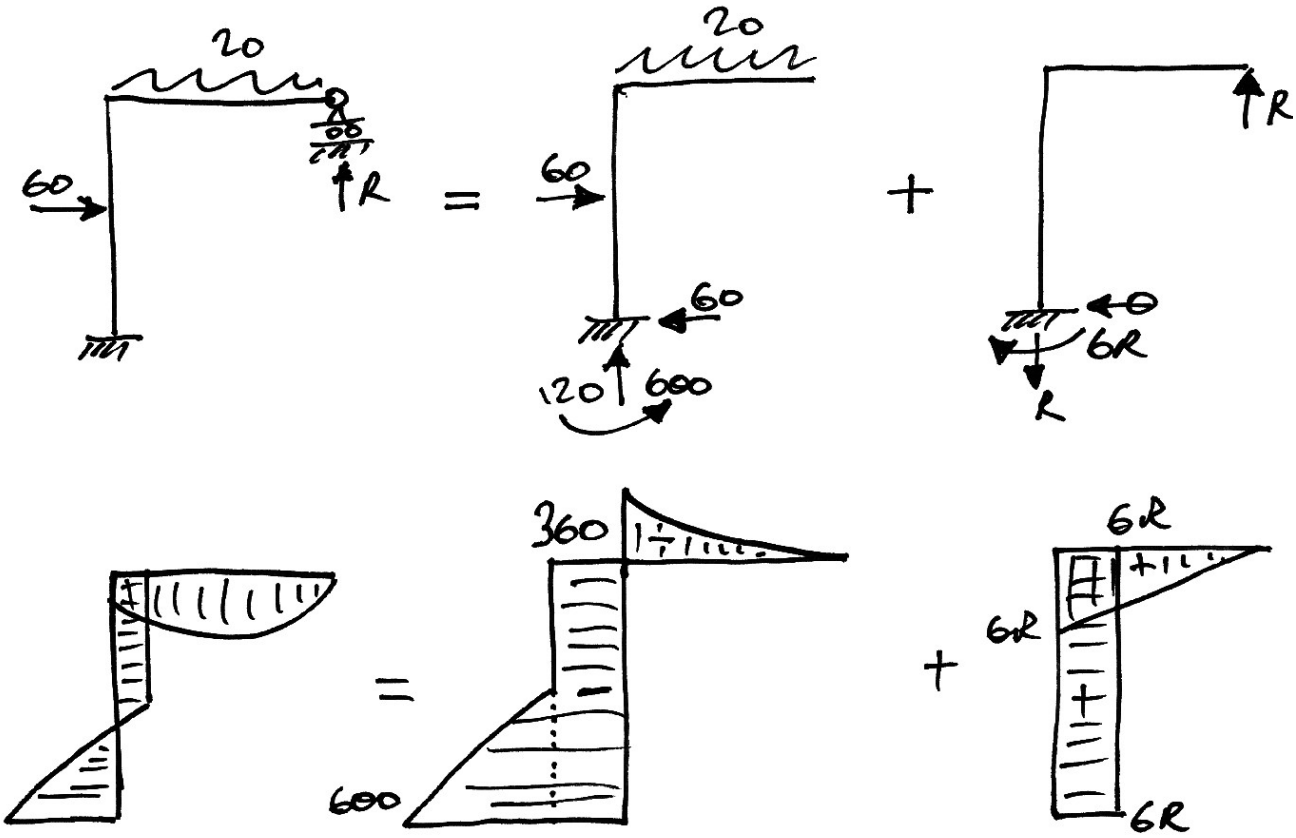


In this frame we have the following added complexities:

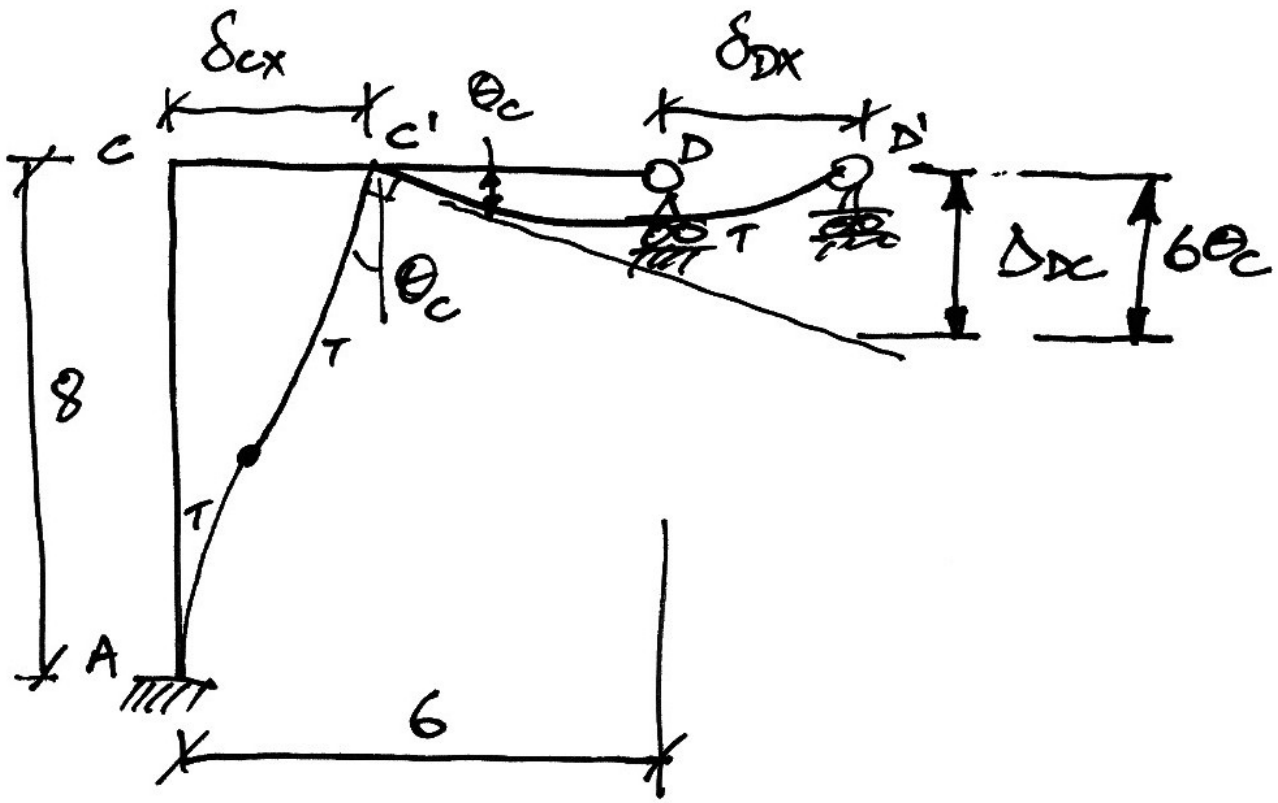
- There is a UDL and a point load which leads to a mix of parabolic, triangular and rectangular BMDs;
- There is a different *EI* value for different parts of the frame – we must take this into account when performing calculations and not just consider the *M* diagram but the M/EI diagram as per Mohr's Theorems.

Solve for a Redundant

As is usual, we split the frame up:



Next we draw the deflected shape diagram of the original structure to identify displacements that we can use:



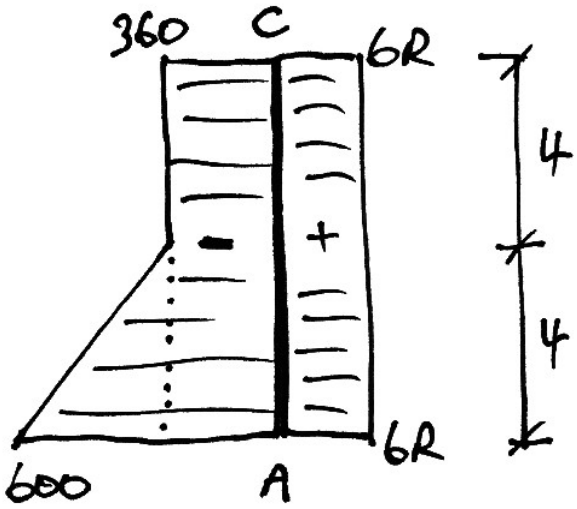
To solve for *R* we will use the vertical intercept Δ_{DC} as shown, because:

- We can determine Δ_{DC} for the original structure in terms of *R* using Mohr II;
- We see that $\Delta_{DC} = 6\theta_C$ and so using Mohr I for the original structure we will find θ_B, again in terms of *R*;
- As usual, we equate the two methods of calculating Δ_{DC} (both are in terms of *R*) and solve for *R*.

The Rotation at *C*

To find the rotation at *C*, we must base our thoughts on the fact that we are only able to calculate the *change in rotation* from one point to another using Mohr I. Thus we identify that we know the rotation at *A* is zero – since it is a fixed support – and we can find the change in rotation from *A* to *C*, using Mohr I. Therefore:

$$\begin{aligned} d\theta_{A\text{ to }C} &= \theta_C - \theta_A \\ &= \theta_C - 0 \\ &= \theta_C \end{aligned}$$

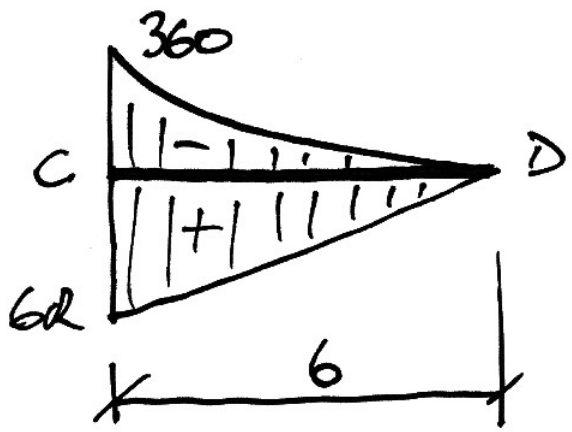


At this point we must recognize that since the frame is swaying to the right, the bending moment on the outside 'dominates' (as we saw for the maximum deflection calculation in Example 6). The change in rotation is the difference of the absolute values of the two diagrams, hence we have, from the figure, and from Mohr I:

$$\begin{aligned} EId\theta_{A\text{ to }C} &= \left|(360\cdot 8) + \left(\frac{1}{2}\cdot 240\cdot 4\right)\right| - \left|(6R\cdot 8)\right| \\ EI\theta_C &= 3360 - 48R \\ \therefore \theta_C &= \frac{3360-48R}{EI} \end{aligned}$$

The Vertical Intercept DC

Using Mohr II and from the figure we have:



$$1.5EI\Delta_{DC} = \left[\left(\frac{1}{2}\cdot 6\cdot 6R\right)\left(\frac{2}{3}\cdot 6\right)\right] - \left[\left(\frac{1}{3}\cdot 6\cdot 360\right)\left(\frac{3}{4}\cdot 6\right)\right]$$

$$1.5EI\Delta_{DC} = 72R - 3240$$

$$\therefore \Delta_{DC} = \frac{48R - 2160}{EI}$$

Note that to have neglected the different *EI* value for member *CD* would change the result significantly.

Solve for *R*

By compatibility of displacement we have $\Delta_{DC} = 6\theta_C$ and so:

$$48R - 2160 = 6(3360 - 48R)$$

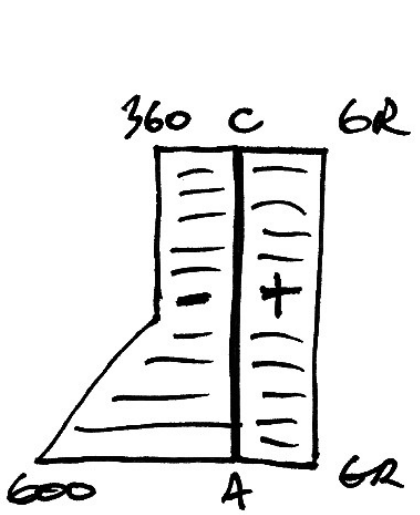
$$336R = 22320$$

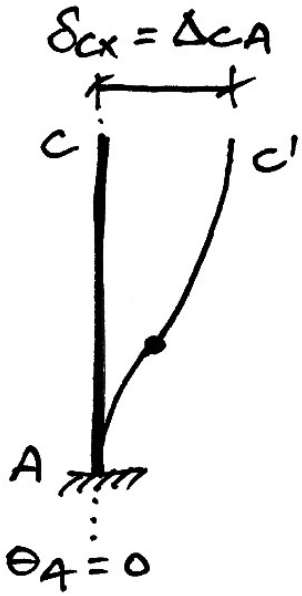
$$R = 66.43 \text{ kN}$$

With *R* now known we can calculate the horizontal deflection at *D*.

Part (b) - Horizontal Deflection at *D*

From the deflected shape diagram of the final frame and by neglecting axial deformation of member *CD*, we see that the horizontal displacement at *D* must be the same as that at *C*. Note that it is easier at this stage to work with the simpler shape of the separate primary and reactant BMDs. Using Mohr II we can find δ_{Cx} as shown:





$$EI\Delta_{CA} = \left[(6R\cdot 8)(4)\right] - \left[(360\cdot 8)(4) + \left(\frac{1}{2}\cdot 4\cdot 240\right)\left(4+\frac{2}{3}\cdot 4\right)\right]$$
$$= 192R - 14720$$

Now substituting $R = 66.4$ kN and $\delta_{Dx} = \delta_{Cx} = \Delta_{BA}$:

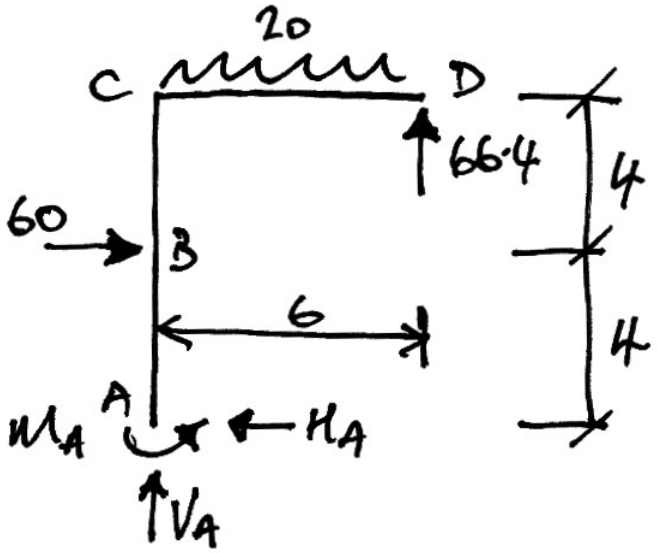
$$\delta_{Dx} = \frac{-1971.2}{EI} = 49.3 \text{ mm} \rightarrow$$

Note that the negative sign indicates that the bending on the outside of the frame dominates, pushing the frame to the right as we expected.

Part (a) – Reactions and Bending Moment Diagram

Reactions

Taking the whole frame, and showing the calculated value for *R*, we have:



$$\sum F_y = 0 \qquad \therefore (20 \cdot 6) - 66.4 - V_A = 0 \qquad \therefore V_A = 53.6 \text{ kN} \uparrow$$

$$\sum F_x = 0 \qquad \therefore H_A - 60 = 0 \qquad \therefore H_A = 60 \text{ kN} \leftarrow$$

$$\sum \text{M about } A = 0 \quad \therefore M_A + 66.4 \cdot 6 - 20 \cdot \frac{6^2}{2} - 60 \cdot 4 = 0 \qquad \therefore M_A = +201.6 \text{ kNm}$$

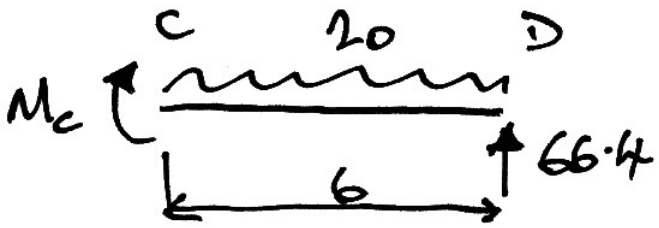
Note that it is easier to use the superposition of the primary and reactant BMDs to find the moment at *A*:

$$M_A = 6(66.4) - 600 = -201.6 \text{ kNm}$$

The negative sign indicate the moment on the outside of the frame dominates and so tension is on the left.

Bending Moment Diagram

We find the moments at salient points:



$$\sum \text{M about } C = 0$$

$$\therefore M_C + 20 \cdot \frac{6^2}{2} - 66.4 \cdot 6 = 0$$

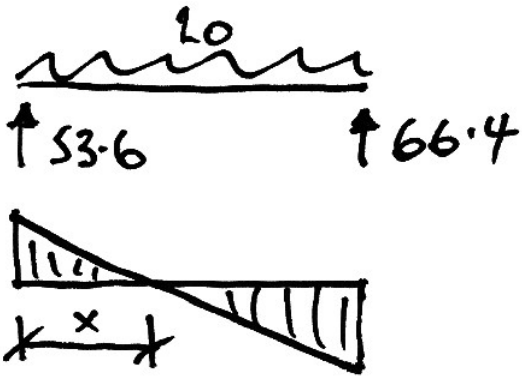
$$\therefore M_C = +38.4 \text{ kNm}$$

And so tension is on the bottom at *C*.

The moment at *B* is most easily found from superposition of the BMDs as before:

$$M_B = 6(66.4) - 360 = 38.4 \text{ kNm}$$

And so tension is on the inside of the frame at *B*. Lastly we must find the value of maximum moment in span *CD*. The position of zero shear is found as:

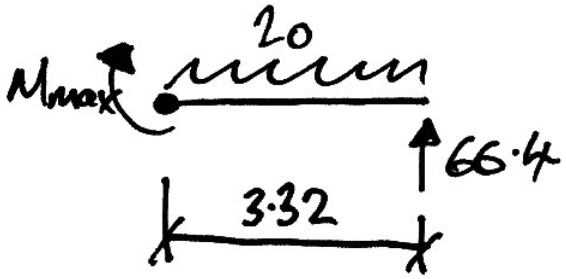


$$x = \frac{53.6}{20} = 2.68 \text{ m}$$

And so the distance from *D* is:

$$6 - 2.68 = 3.32 \text{ m}$$

The maximum moment is thus found from a free body diagram as follows:



$$\sum \text{M about } X = 0$$

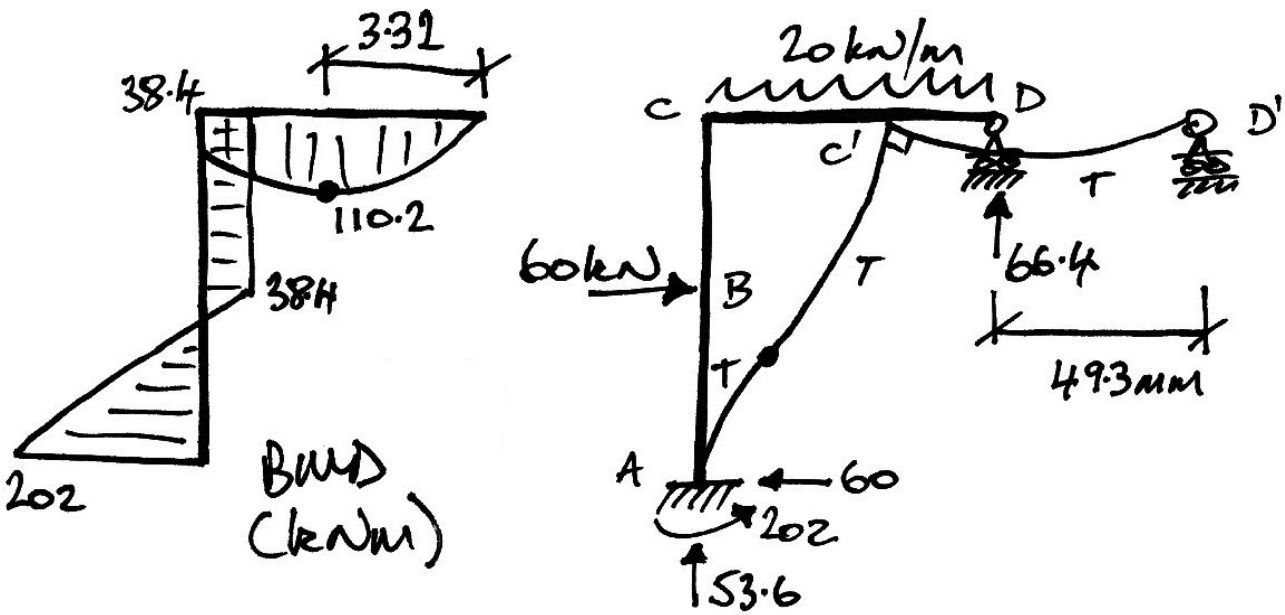
$$\therefore M_{\max} + 20 \cdot \frac{3.32^2}{2} - 66.4 \cdot 3.32 = 0$$

$$\therefore M_C = +110.2 \text{ kNm}$$

And so tension is on the bottom as expected.

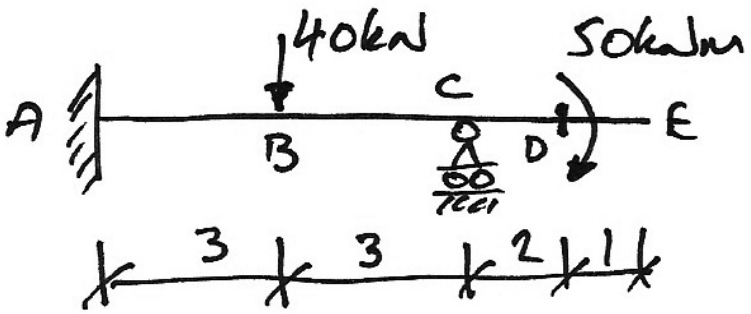
Summary of Solution

In summary the final solution for this frame is:



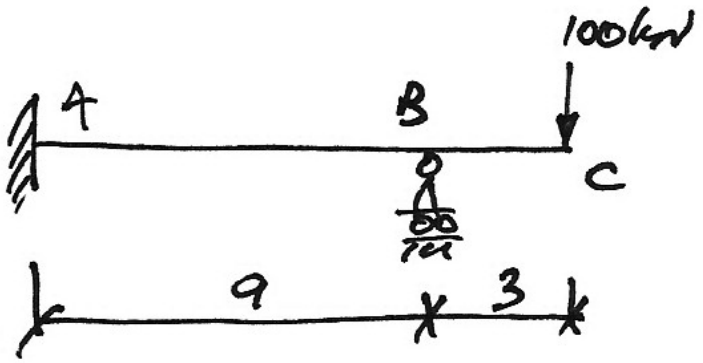
4.6 Problems

1. For the following prismatic beam, find the bending moment diagram and the rotation at *E* in terms of *EI*.



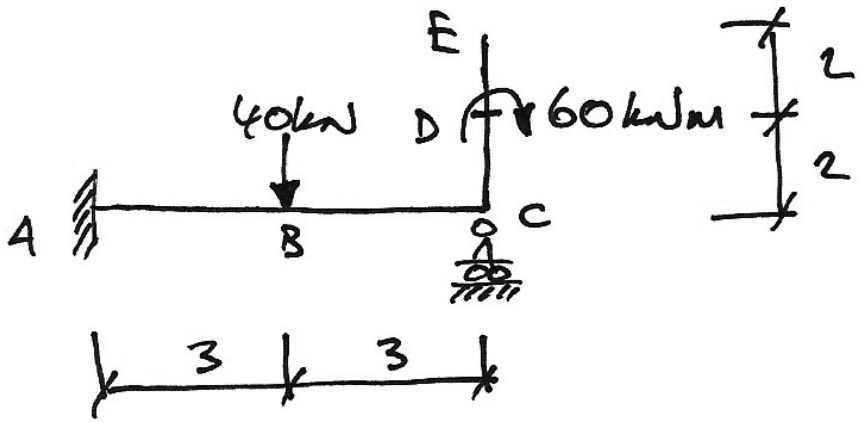
Ans. $V_C = 25\,\text{kN}\uparrow$, $\theta_E = 130/EI$

2. For the following prismatic beam, find the bending moment diagram and the rotation at *C* in terms of *EI*.



Ans. $V_C = 150\,\text{kN}\uparrow$, $\theta_E = 1125/EI$

3. For the following prismatic frame, find the bending moment and shear force diagrams and the horizontal deflection at *E* in terms of *EI*.



Ans. $V_C = 27.5\,\text{kN}\uparrow$, $\delta_{Ex} = 540/EI$, $\theta_C = 45/EI$

5. Further Developments

5.1 Theorem of Three Moments

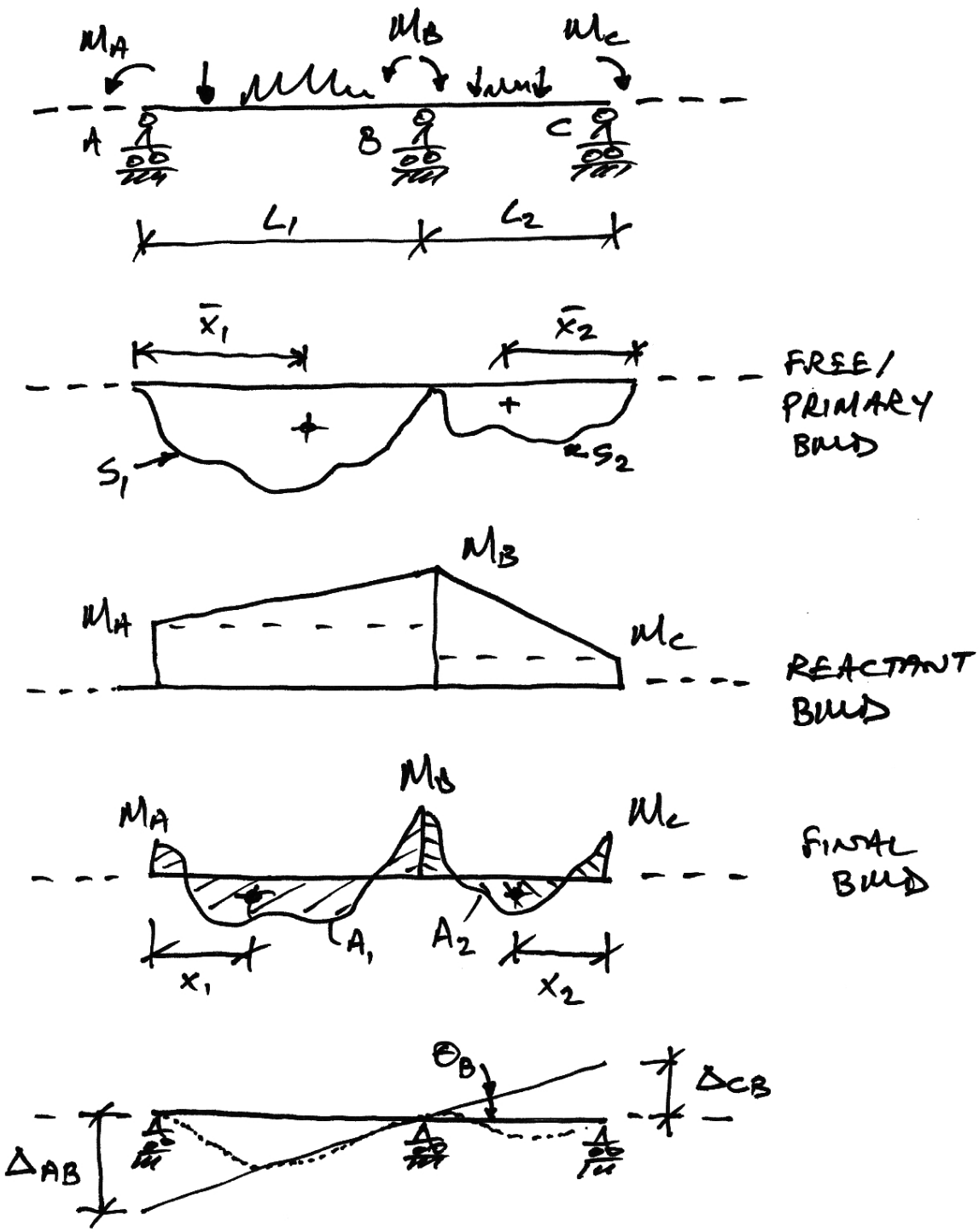
Introduction

Continuous beams feature in many structures, and as such the ability to analyse them is crucial to efficient design. Clapeyron derived the Three Moment Theorem in about 1857 for this purpose, thereby enabling the design of the previously ‘undesignable’. He derived them using Mohr’s Theorems.

They were initially derived for the very general case of spans with different flexural rigidities, span lengths and support levels. We will only consider the case of different span lengths to keep the problem simple. Even so, the solution procedure is exactly the same and the result is widely applicable.

Development

We consider the following two internal spans of an arbitrary continuous beam:



To solve the problem we will identify two relationships and join them to form an equation that enables us to solve continuous beams.

First, we calculate the two vertical intercepts, Δ_{AB} and Δ_{CB}:

$$\Delta_{AB} = \frac{A_1 x_1}{EI} = \theta_B L_1 \tag{1}$$

$$\Delta_{CB} = \frac{A_2 x_2}{EI} = -\theta_B L_2 \tag{2}$$

Note that Δ_{CB} is negative since it is upwards. We can solve these two equations for θ_B:

$$\frac{A_1 x_1}{EIL_1} = \theta_B$$

$$\frac{A_2 x_2}{EIL_2} = -\theta_B$$

And then add them to get:

$$\frac{A_1 x_1}{EIL_1} + \frac{A_2 x_2}{EIL_2} = 0$$

And since *EI* is common we have our first relationship:

$$\frac{A_1 x_1}{L_1} + \frac{A_2 x_2}{L_2} = 0 \tag{3}$$

The next step involves determining the first moment of area of the two final bending moment diagrams in terms of the free and reactant bending moment diagrams. In words, the first moment of the final BMD about *A* is equal to the sum of the first moments of the free BMD and reactant BMDs about *A*. Mathematically, from the figure, we thus have:

$$A_1 x_1 = S_1 \overline{x}_1 + \left[\left(M_A L_1 \right) \left(\frac{L_1}{2} \right) + \left(\frac{1}{2} \left(M_B - M_A \right) L_1 \right) \left(\frac{2}{3} L_1 \right) \right] \tag{4}$$

In which the reactant BMD has been broken into a rectangular and triangular parts (dotted in the figure). Similarly, we have:

$$A_2 x_2 = S_2 \overline{x}_2 + \left[\left(M_C L_2 \right) \left(\frac{L_2}{2} \right) + \left(\frac{1}{2} \left(M_B - M_C \right) L_2 \right) \left(\frac{2}{3} L_2 \right) \right] \tag{5}$$

Introducing equations (4) and (5) into equation (3) gives:

$$\left[\frac{S_1 \overline{x}_1}{L_1} + \frac{M_A L_1}{2} + \frac{\left(M_B - M_A \right) L_1}{3} \right] + \left[\frac{S_2 \overline{x}_2}{L_2} + \frac{M_C L_2}{2} + \frac{\left(M_B - M_C \right) L_2}{3} \right] = 0$$

Carrying out the algebra:

$$\left(\frac{M_A L_1}{2} + \frac{M_B L_1}{3} - \frac{M_A L_1}{3} \right) + \left(\frac{M_C L_2}{2} + \frac{M_B L_2}{3} - \frac{M_C L_2}{3} \right) = -\left(\frac{S_1 \overline{x}_1}{L_1} + \frac{S_2 \overline{x}_2}{L_2} \right)$$

$$\left(\frac{M_A L_1}{6} + \frac{2 M_B L_1}{6} \right) + \left(\frac{M_C L_2}{6} + \frac{2 M_B L_2}{6} \right) = -\left(\frac{S_1 \overline{x}_1}{L_1} + \frac{S_2 \overline{x}_2}{L_2} \right)$$

And finally we arrive at the **Three Moment Equation**:

$$M_A L_1 + 2M_B\left(L_1 + L_2\right) + M_C L_2 = -6\left(\frac{S_1\overline{x}_1}{L_1} + \frac{S_2\overline{x}_2}{L_2}\right) \tag{6}$$

This equation relates the unknown reactant moments to the free bending moment diagram for each two spans of a continuous beam. By writing this equation for each adjacent pair of spans, a sufficient number of equations to solve for the unknown reactant moments result.

The term in brackets on the right of the equation represents the total angular discontinuity ($EI\left(\theta_{BA} + \theta_{BC}\right)$) at *B* if *A*, *B* and *C* were pinned supports.

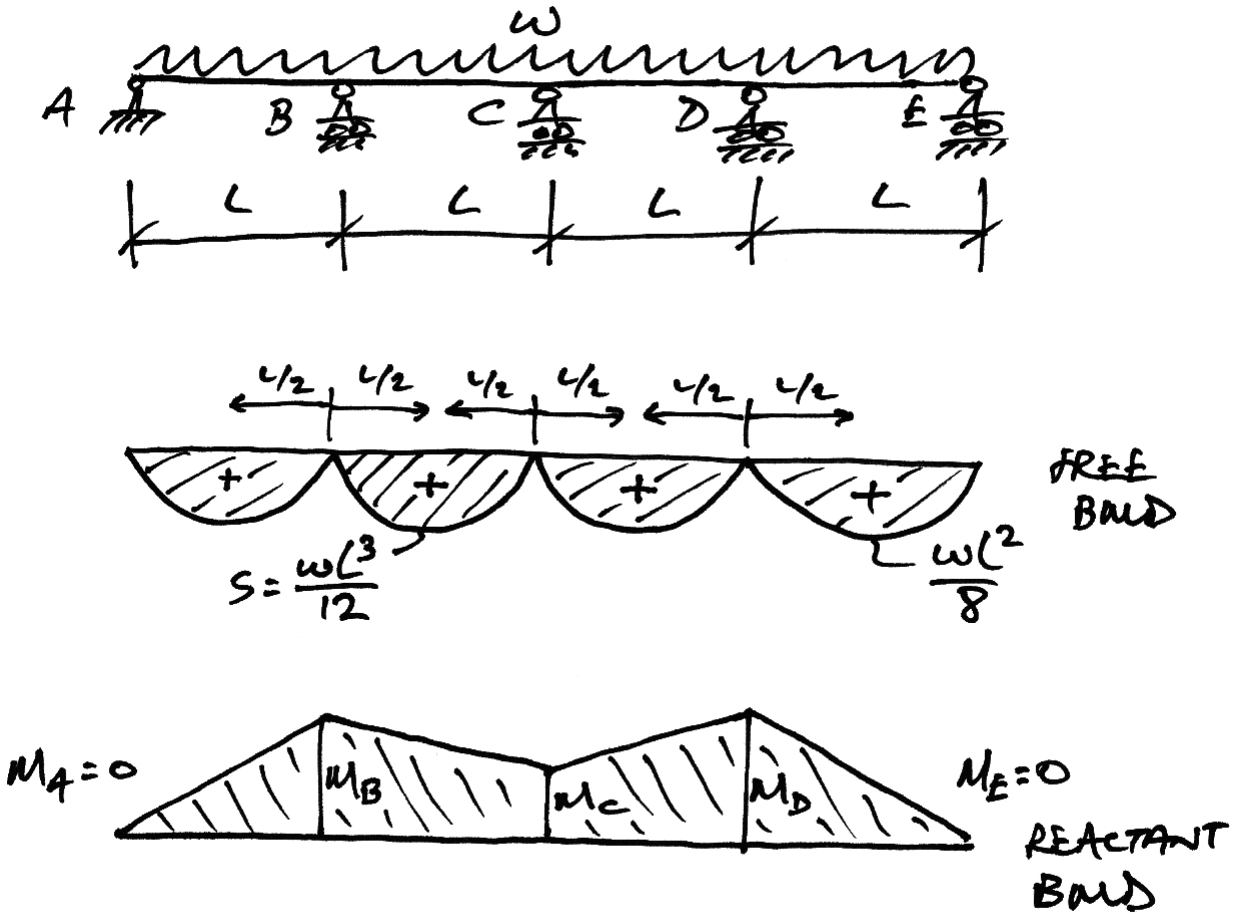
As a further development, we can use equations (1) and (2) with Mohr's First Theorem to find:

$$\begin{aligned}
\theta_A &= \frac{1}{EI}\left[S_1\left(1 - \frac{\overline{x}_1}{L_1}\right) - \frac{L_1}{6}\left(2M_A + M_B\right)\right] \\
\theta_B &= \frac{1}{EI}\left[-S_1\frac{\overline{x}_1}{L_1} + \frac{L_1}{6}\left(M_A + 2M_B\right)\right] \\
&= \frac{1}{EI}\left[-S_2\frac{\overline{x}_2}{L_2} + \frac{L_2}{6}\left(M_C + 2M_B\right)\right] \\
\theta_C &= \frac{1}{EI}\left[S_2\left(1 - \frac{\overline{x}_2}{L_2}\right) - \frac{L_2}{6}\left(2M_C + M_B\right)\right]
\end{aligned} \tag{7}$$

With this information, all deflections along the beam can be found using the numerical procedure to be explained later.

Example 10

To illustrate the application of the Three Moment Theorem, and also to derive a useful result, we will consider the general case of a four-span beam with equal spans, *L*, subject to a UDL, *w*:



In the figure, the areas of the free BMDs are all:

$$S_{1,2,3,4} = \frac{2}{3}\left(\frac{wL^2}{8}\right)L = \frac{wL^3}{12}$$

And the distances to the centroids are all $L/2$. Thus we can write:

$$\frac{S\overline{x}}{L} = \frac{1}{L}\left(\frac{wL^3}{12}\right)\left(\frac{L}{2}\right) = \frac{wL^3}{24}$$

Next, we apply the Three Moment Equation to each pair of spans:

ABC: $$2M_B\left(L+L\right) + M_C L = -6\left(\frac{wL^3}{24} + \frac{wL^3}{24}\right)$$

BCD: $$M_B L + 2M_C\left(L+L\right) + M_D L = -6\left(\frac{wL^3}{24} + \frac{wL^3}{24}\right)$$

CDE: $$M_C L + 2M_D\left(L+L\right) = -6\left(\frac{wL^3}{24} + \frac{wL^3}{24}\right)$$

Simplifying:

$$4M_B + M_C = -\frac{wL^2}{2}$$
$$M_B + 4M_C + M_D = -\frac{wL^2}{2}$$
$$M_C + 4M_D = -\frac{wL^2}{2}$$

This is three equations with three unknowns and is thus readily solvable.

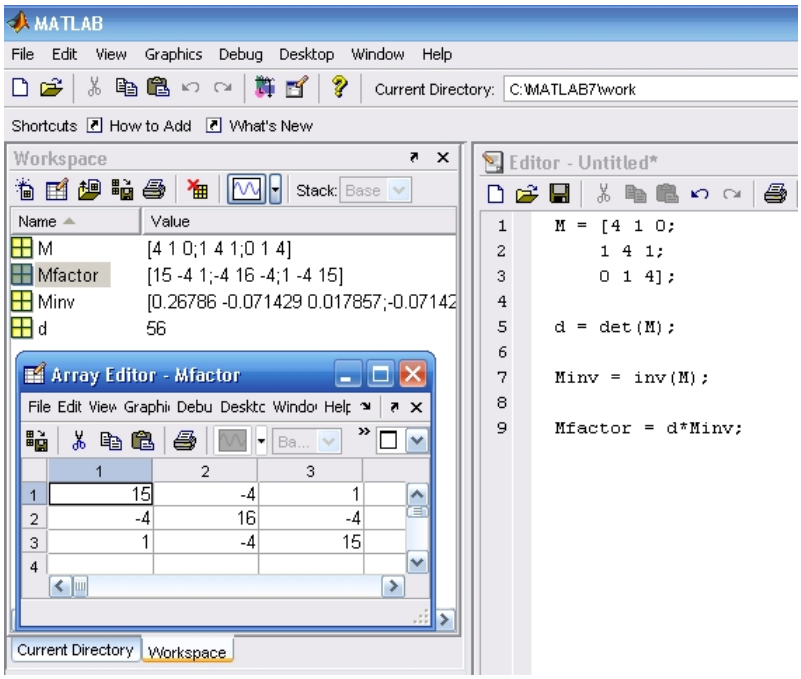
An algebraic approach is perfectly reasonable, but we can make better use fo the tools at our disposal if we rewrite this in matrix form:

$$\begin{bmatrix} 4 & 1 & 0 \\ 1 & 4 & 1 \\ 0 & 1 & 4 \end{bmatrix} \begin{Bmatrix} M_B \\ M_C \\ M_D \end{Bmatrix} = -\frac{wL^2}{2} \begin{Bmatrix} 1 \\ 1 \\ 1 \end{Bmatrix}$$

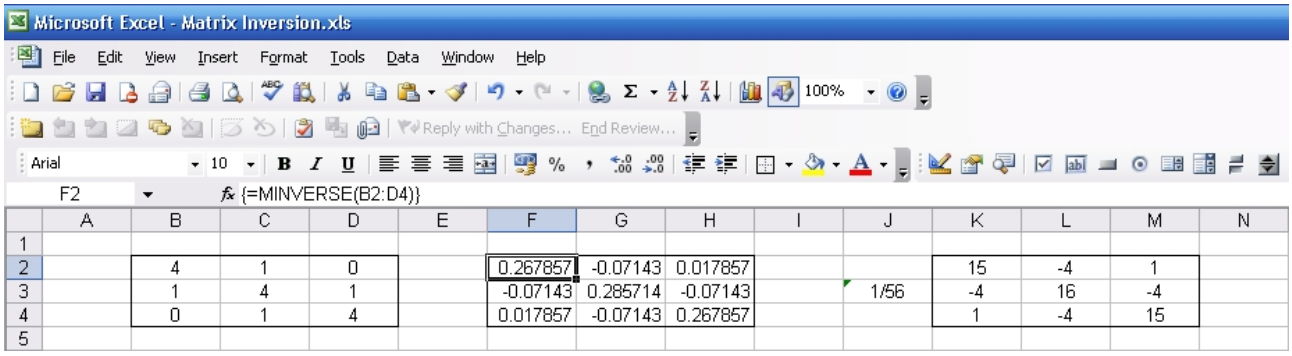
Now we can write can solve for the moment vector suing matrix inversion:

$$\begin{Bmatrix} M_B \\ M_C \\ M_D \end{Bmatrix} = -\frac{wL^2}{2} \begin{bmatrix} 4 & 1 & 0 \\ 1 & 4 & 1 \\ 0 & 1 & 4 \end{bmatrix}^{-1} \begin{Bmatrix} 1 \\ 1 \\ 1 \end{Bmatrix}$$

To obtain the inverse of the 3×3 matrix we could resort to algebra, but a better idea is to use Excel or Matlab. Using Matlab:



We can see that the decimal results of the matrix inverse are really just multiples of 1/56 (the matrix determinant). In Excel we can find the matrix inverse (using MINVERSE), but cannot find the determinant directly, so we use the Matlab result:



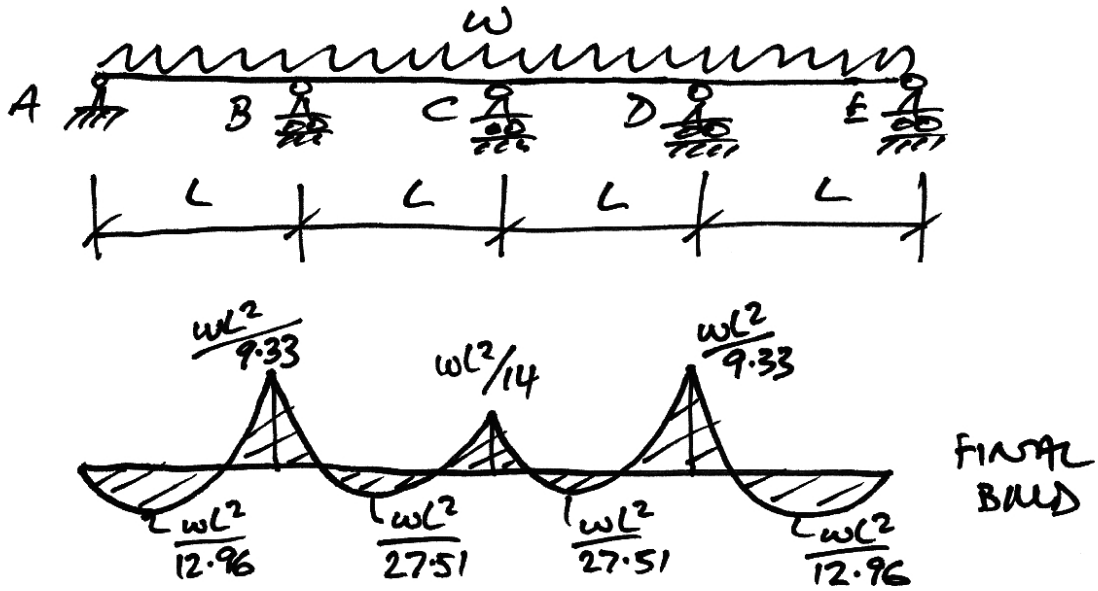
	A	B	C	D	E	F	G	H	I	J	K	L	M	N
1														
2		4	1	0		0.267857	-0.07143	0.017857			15	-4	1	
3		1	4	1		-0.07143	0.285714	-0.07143		1/56	-4	16	-4	
4		0	1	4		0.017857	-0.07143	0.267857			1	-4	15	
5														

Thus our solution becomes:

$$\begin{Bmatrix} M_B \\ M_C \\ M_D \end{Bmatrix} = -\frac{wL^2}{2}\left(\frac{1}{56}\right)\begin{bmatrix} 15 & -4 & 1 \\ -4 & 16 & -4 \\ 1 & -4 & 15 \end{bmatrix}\begin{Bmatrix} 1 \\ 1 \\ 1 \end{Bmatrix} = -\frac{wL^2}{112}\begin{Bmatrix} 12 \\ 8 \\ 12 \end{Bmatrix}$$

It is quite useful to know the denominators for easier comparisons, thus:

$$\begin{bmatrix} M_B & M_C & M_D \end{bmatrix} = \begin{bmatrix} \frac{-wL^2}{9.33} & \frac{-wL^2}{14} & \frac{-wL^2}{9.33} \end{bmatrix}$$



5.2 Numerical Calculation of Deformation

Introduction

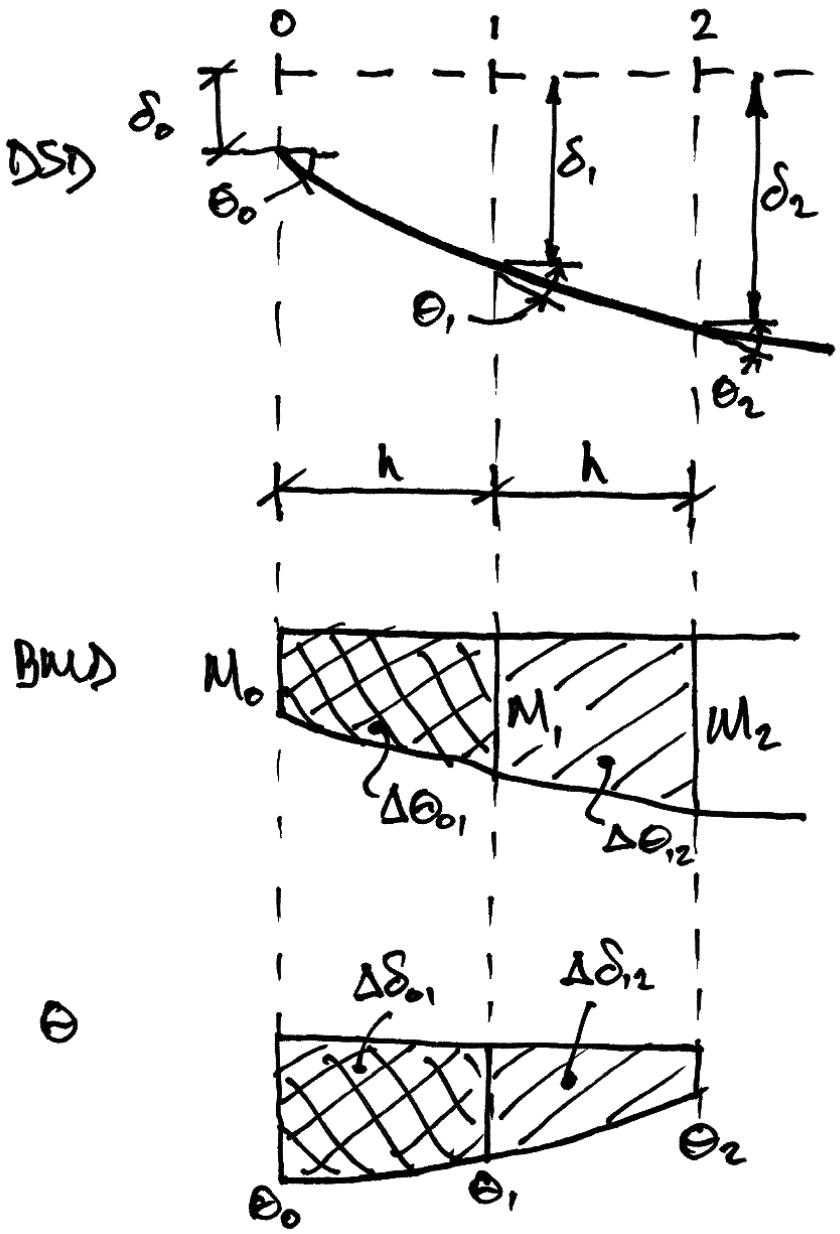
One of the main applications of the Moment-Area method in the modern structural analysis environment, where use of computers is prevalent, is in the calculation of displacements. Most structural analysis software is based on the matrix stiffness (or finite element) method. This analysis procedure returns the displacements and rotations at node points only. The problem then remains to determine the displacements along members, in between the nodes. This is where the moment-area method is applied in typical analysis software programs.

We will demonstrate a simple procedure to find the deflections and rotations along a member once the bending moments are known at discrete points along the member. In addition, we will consider the member prismatic: *EI* will be taken as constant, though this not need be so in general.

You can download all the files and scripts from the course website.

Development

Consider a portion of a deformed member with bending moments known:



Our aim is to determine the rotation and deflection at each station (1, 2, …) given the values of bending moment $M_1, M_2, \ldots$ and the starting rotation and deflection, θ_0, δ_0. We base our development on the fundamental Euler-Bernoulli relationships upon which Mohr's Theorems were developed:

$$\theta_i - \theta_{i-1} = \int_{i-1}^{i} \frac{M}{EI} dx \tag{8}$$

$$\delta_i - \delta_{i-1} = \int_{i-1}^{i} \theta dx \tag{9}$$

From these equations, and from the diagram, we can see that:

$$\Delta\theta = \frac{1}{EI} \int M dx$$

$$\Delta\delta = \int \theta dx$$

Thus we have:

$$\theta_i = \theta_{i-1} + \Delta\theta = \theta_{i-1} + \frac{1}{EI} \int M dx \tag{10}$$

$$\delta_i = \delta_{i-1} + \Delta\delta = \delta_{i-1} + \int \theta dx \tag{11}$$

In this way once θ_0, δ_0 are known, we can proceed along the member establishing rotations and deflections at each point.

Implementation

To implement this method, it just remains to carry out the integrations. To keep things simple, we will use the Trapezoidal Rule. More accurate methods are possible, such as Simpson's Rule, but since we can usually choose the number of stations to be large, little error will result.

Thus equations (10) and (11) become:

$$\theta_i = \theta_{i-1} + \frac{1}{2}\left(M_{i-1} + M_i\right)\frac{h}{EI} \tag{12}$$

$$\delta_i = \delta_{i-1} + \frac{1}{2}\left(\theta_{i-1} + \theta_i\right)h \tag{13}$$

To proceed we will consider the following example, for which we know the result:

- Simply-support beam;
- $L = 6\ \text{m}$;
- $EI = 180 \times 10^3\ \text{kNm}^2$;
- Loading: UDL of 20 kN/m.

Our main theoretical result is:

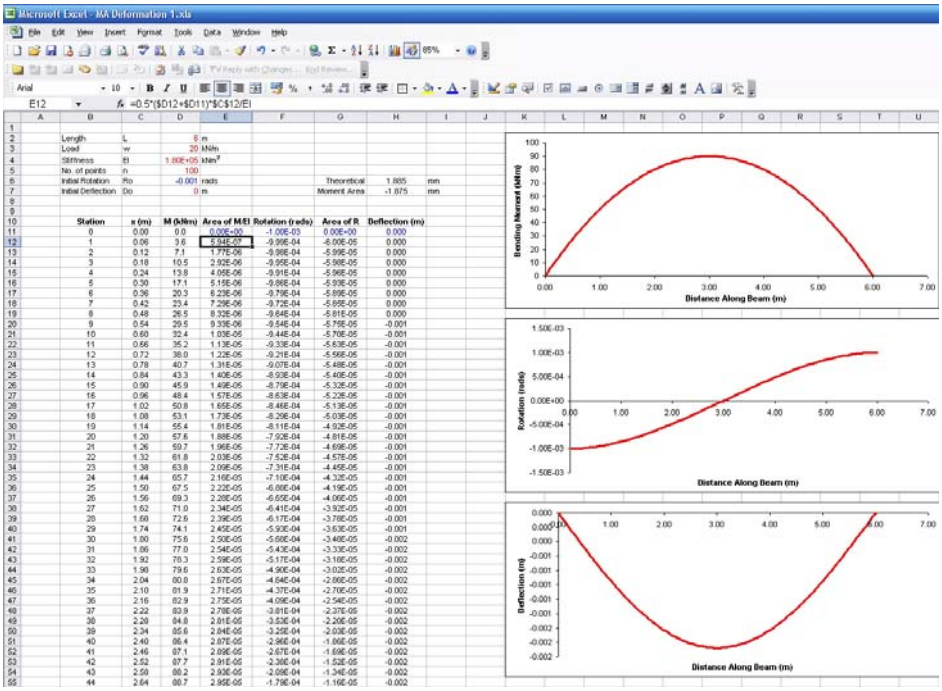
$$\delta = \frac{5wL^4}{384EI} = \frac{5(20)6^4}{384\left(1.8 \times 10^5\right)}\frac{1}{1000} = 1.885\,\text{mm}$$

MS Excel

We can implement the formulas as follows:

Length	L	6	m			
Load	w	20	kN/m			
Stiffness	EI	180000	kNm^2			
No. of points	n	100				
Initial Rotation	Ro	=-w*L^3/(24*EI)	rads			
Initial Deflection	Do	0	m			
Station	**x (m)**	**M (kNm)**	**Area of M/EI**	**Rotation (rads)**	**Area of R**	**Deflection (m)**
0	=(B11/n)*L	=w*L*$C11/2-w*$C11^2/2	0	=Ro	0	=Di
1	=(B12/n)*L	=w*L*$C12/2-w*$C12^2/2	=0.5*($D12+$D11)*$C$12/EI	=F11+E12	=0.5*($F12+$F11)*$C$12	=G12+H11
2	=(B13/n)*L	=w*L*$C13/2-w*$C13^2/2	=0.5*($D13+$D12)*$C$12/EI	=F12+E13	=0.5*($F13+$F12)*$C$12	=G13+H12
3	=(B14/n)*L	=w*L*$C14/2-w*$C14^2/2	=0.5*($D14+$D13)*$C$12/EI	=F13+E14	=0.5*($F14+$F13)*$C$12	=G14+H13
4	=(B15/n)*L	=w*L*$C15/2-w*$C15^2/2	=0.5*($D15+$D14)*$C$12/EI	=F14+E15	=0.5*($F15+$F14)*$C$12	=G15+H14

And drag down these formulas for 100 points to get the following spreadsheet:



The deflection with 100 points along the beam is 1.875 mm – a very slight difference to the theoretical result.

Matlab

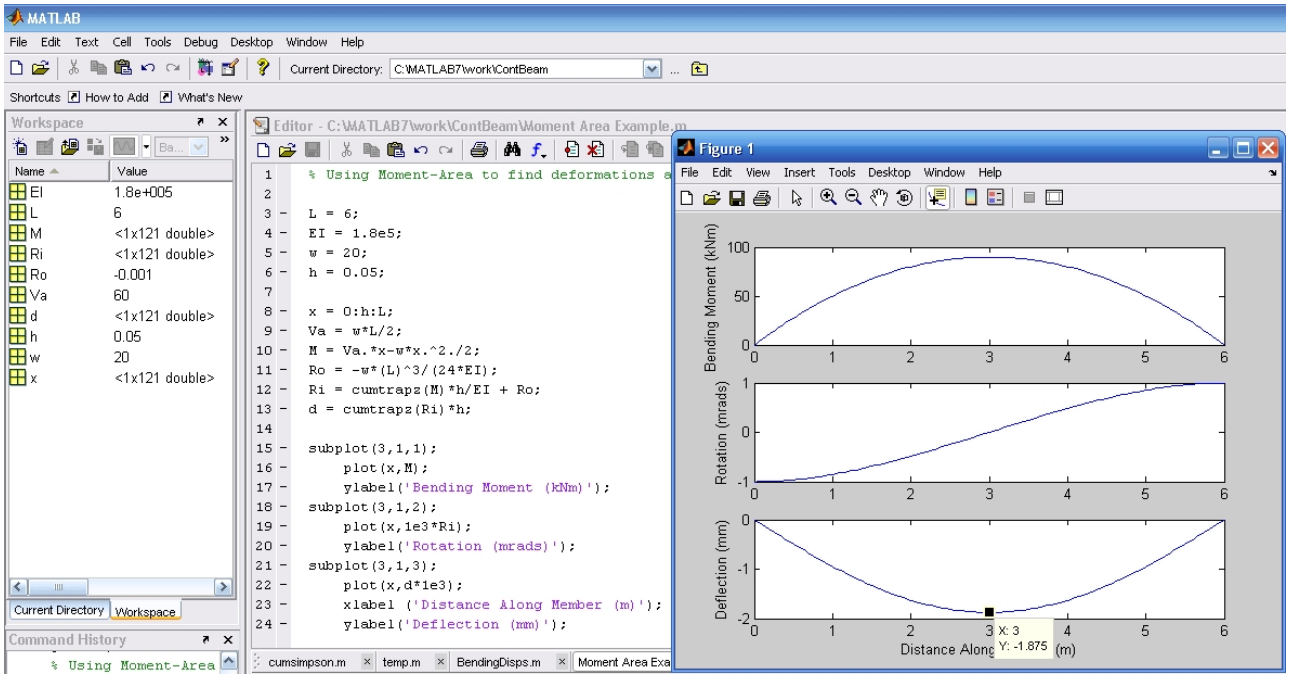
Matlab has a very useful function for our purposes here: `cumtrapz`. This function returns the cumulative results of trapezoidal integration along a function. Thus our script becomes:

```
% Using Moment-Area to find deformations along members

L = 6;
EI = 1.8e5;
w = 20;
h = 0.05;

x = 0:h:L;
Va = w*L/2;
M = Va.*x-w*x.^2./2;
Ro = -w*(L)^3/(24*EI);
Ri = cumtrapz(M)*h/EI + Ro;
d = cumtrapz(Ri)*h;

subplot(3,1,1);
    plot(x,M);
    ylabel('Bending Moment (kNm)');
subplot(3,1,2);
    plot(x,1e3*Ri);
    ylabel('Rotation (mrads)');
subplot(3,1,3);
    plot(x,d*1e3);
    xlabel ('Distance Along Member (m)');
    ylabel('Deflection (mm)');
```

As may be seen, most of this script is to generate the plots. The `cumtrapz` function takes the hard work out of this approach.

The central deflection result is 1.875 mm again.

The output is in this screenshot:



For maximum flexibility, it is better to write a generic function to perform these tasks:

```
function [R d] = MomentArea(M, EI, h, Ro, do)
% This function calculates the rotations and deflections along a
flexural
% member given the bending moment vector, M, a distance step, h,
initial
% deflection and rotation at node 0, do, Ro, and the flexural
rigidity, EI.

n = length(M);                    % number of stations
R = zeros(n,1);                   % vector of rotations
d = zeros(n,1);                   % vector of deflections

R(1) = Ro;                        % assign starting rotation and
deflection
d(1) = do;

R = cumtrapz(M)*h/EI + Ro;      % Do moment area calcs
d = cumtrapz(R)*h;
```

To use this function for our example we make the following calls:

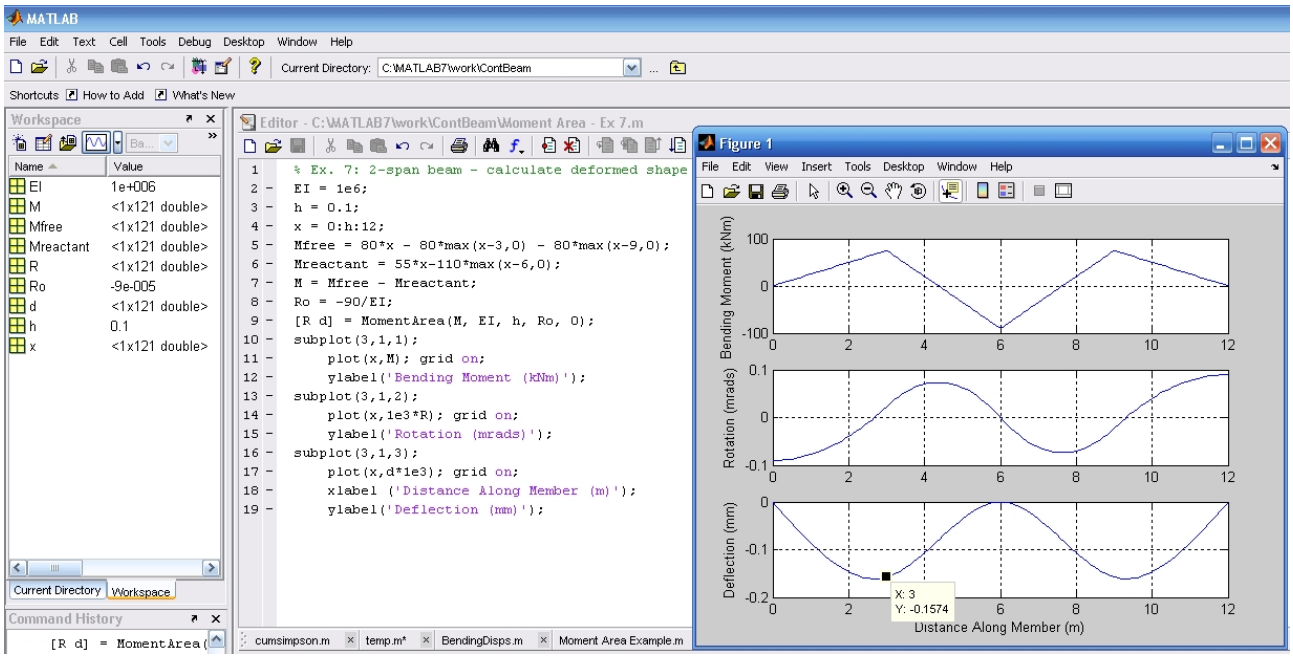
```
L = 6;
EI = 1.8e5;
w = 20;
h = 0.05;
x = 0:h:L;
Va = w*L/2;
M = Va.*x-w*x.^2./2;
Ro = -w*(L)^3/(24*EI);
[R d] = MomentArea(M, EI, h, Ro, 0);
```

And once again, of course, our result is 1.875 mm.

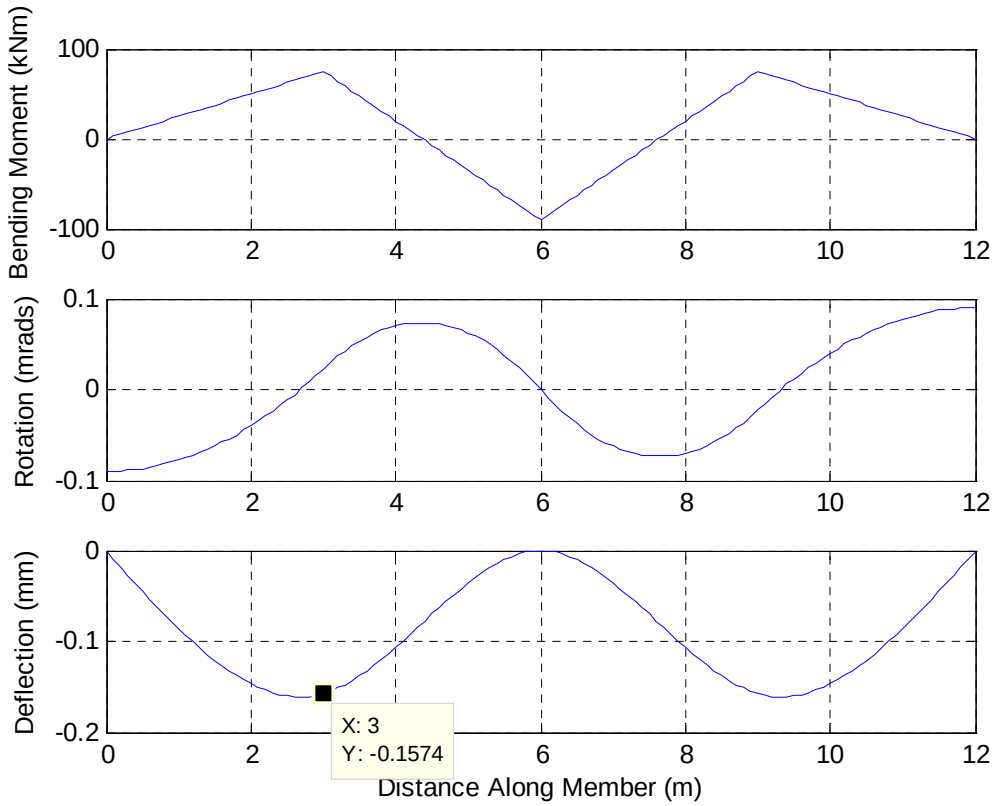
As one final example, we calculate deflections for the beam of Example 7. To do this we make use of the calculated value for $\theta_A = -90/EI$ and use the following script:

```
% Ex. 7: 2-span beam - calculate deformed shape
EI = 1e6;
h = 0.1;
x = 0:h:12;
Mfree = 80*x - 80*max(x-3,0) - 80*max(x-9,0);
Mreactant = 55*x-110*max(x-6,0);
M = Mfree - Mreactant;
Ro = -90/EI;
[R d] = MomentArea(M, EI, h, Ro, 0);
subplot(3,1,1);
    plot(x,M); grid on;
    ylabel('Bending Moment (kNm)');
subplot(3,1,2);
    plot(x,1e3*R); grid on;
    ylabel('Rotation (mrads)');
subplot(3,1,3);
    plot(x,d*1e3); grid on;
    xlabel ('Distance Along Member (m)');
    ylabel('Deflection (mm)');
```

Note that we have used a value of *EI* that makes it easy to interpret the results in terms of *EI*.



As can be seen, the complete deflected profile is now available to us. Further, the deflection at D is found to be $157.4/EI$, which compares well to the theoretical value of $157.5/EI$, found in Example 7.



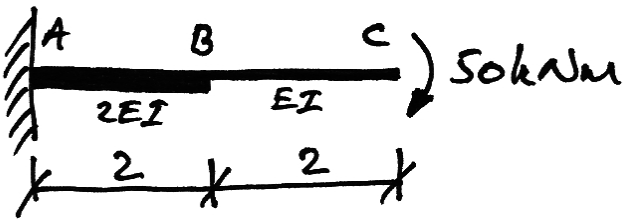
5.3 Non-Prismatic Members

Introduction

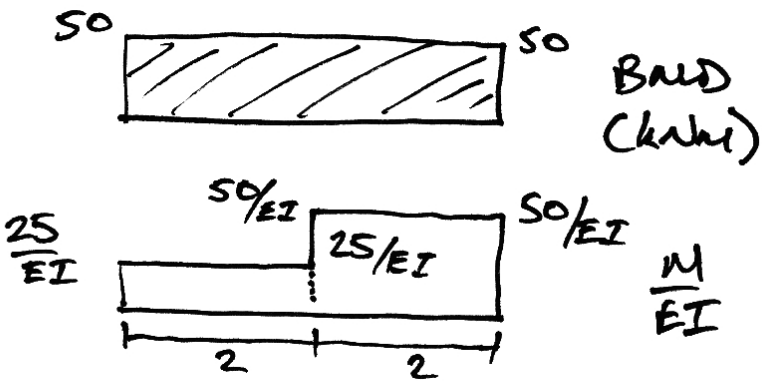
In all examples so far we have only considered members whose properties do not change along their length. This is clearly quite a simplification since it is necessary for maximum structural efficiency that structures change shape to deal with increasing or reducing bending moments etc. The Moment-Area Method is ideally suited to such analyses. We will consider one simple example, and one slightly more complex and general.

Example 11

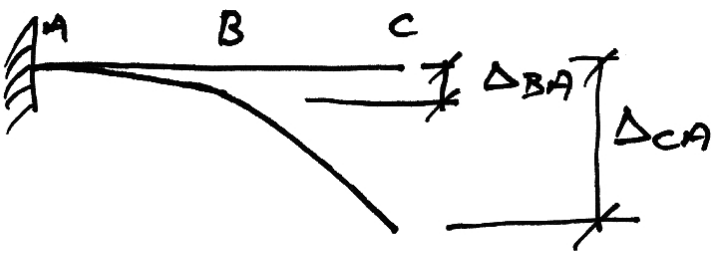
We consider the following cantilever and determine the deflections at *B* and *C*:



The BMD and curvature diagrams thus become:



To calculate the deflections, consider the deflected shape diagram:



From Mohr's First Theorem:

$$\delta_B = \Delta_{BA} = \left[\frac{25}{EI} \cdot 2\right][1]$$

Thus:

$$\delta_B = \frac{50}{EI}$$

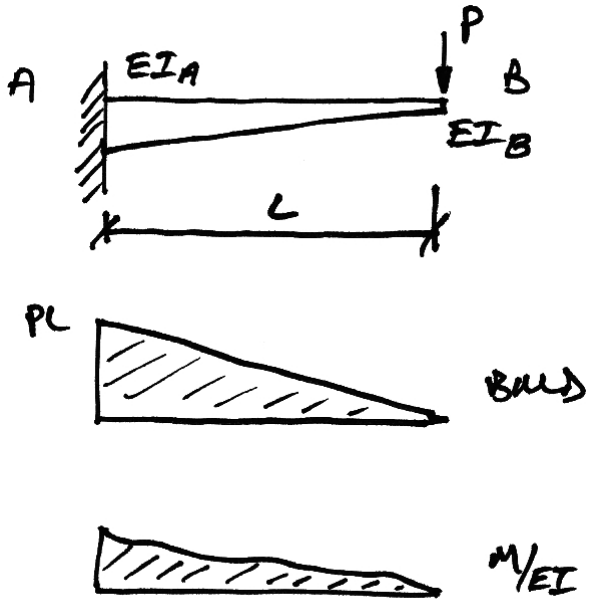
Similarly, though with more terms for the deflection at *C* we have:

$$\delta_C = \Delta_{CA} = \left[\frac{25}{EI} \cdot 2\right][3] + \left[\frac{50}{EI} \cdot 2\right][1]$$

$$\delta_C = \frac{250}{EI}$$

Example 12

We determine here an expression for the deflection at the end of a cantilever subject to a point load at its tip which has linearly varying flexural rigidity:



We must derive expression for both the moment and the flexural rigidity. Considering the coordinate *x*, increasing from zero at *B* to *L* at *A*:

$$M(x) = Px$$

$$EI(x) = EI_B + (EI_A - EI_B)\frac{x}{L}$$

If we introduce the following measure of the increase in *EI*:

$$k = \frac{EI_A - EI_B}{EI_B}$$

We can write:

$$EI(x) = EI_B\left(1 + k\frac{x}{L}\right)$$

Now we can write the equation for curvature:

$$\frac{M}{EI}(x) = \frac{Px}{EI_B\left(1 + k\frac{x}{L}\right)}$$
$$= \frac{PL}{EI_B} \cdot \frac{x}{L + kx}$$

To find the tip deflection we write:

$$\delta_B = \Delta_{BA} = \int_0^L \frac{M}{EI}(x)\, x\, dx$$

And solving this (using symbolic computation!) gives:

$$\delta_B = \frac{PL^3}{2EI}\left[\frac{-2k + 2\log(1+k) + k^2}{k^3}\right]$$

To retrieve our more familiar result for a prismatic member, we must use L'Hopital's Rule to find the limit as $k \to 0$. As may be verified by symbolic computation:

$$\delta_B\left|\text{Prismatic}\right| = \lim_{k\to 0}\frac{PL^3}{2EI}\left[\frac{-2k + 2\log(1+k) + k^2}{k^3}\right] = \frac{PL^3}{3EI}$$

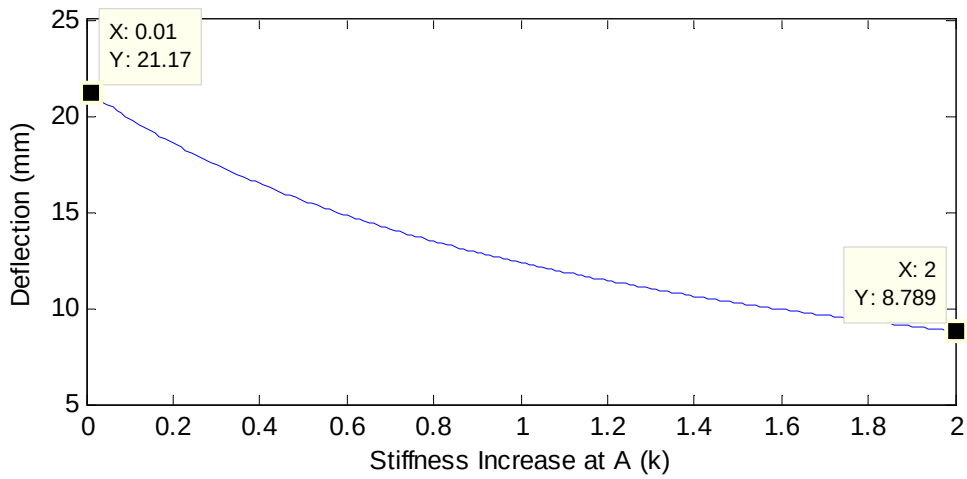
As a sample application, let's take the following parameters:

- $P = 10\ \text{kN}$;
- $L = 4\ \text{m}$;
- $EI_B = 10\ \text{MNm}^2$.

We will investigate the change in deflection with the increase in *EI* at *A*. Firstly, we find our prismatic result:

$$\delta_B\left|\text{Prismatic}\right| = \frac{PL^3}{3EI} = \frac{10\left(4^3\right)}{3\left(10\times10^3\right)} = 21.33\ \text{mm}$$

And then we plot the deflection for a range of *k* values:



As can be seen, when $k = 2$, in other words when $EI_A = 3EI_B$, our deflection is 8.79 mm – a reduction to 41% of the prismatic deflection.

Matlab Scripts

The Matlab scripts to calculate the previous results are:

```
% Use Symbolic Toolbox to perform integration
syms P L EI x k positive;

M = sym('P*x');              % M equation
EI = sym('EI*(1+k*x/L)');    % EI equation
Mohr2 = M/EI*x;              % 1st moment of M/EI diagram
def = int(Mohr2,x,0,L);      % definite integral
pretty(def);                 % display result

limit(def,k,0);              % Prove limit as k->0 is prismatic
result

% Plot change in deflection by varying k
clear all;
P = 10; L = 4; EI = 10e3; k = 0.01:0.01:2;
d = 1/2*P*L^3*(-2.*k+2*log(1+k)+k.^2)/EI./k.^3;
plot(k,d*1e3); xlabel('Stiffness Increase at A (k)');
ylabel('Deflection (mm)')

d1 = P*L^3/(3*EI);           % prismatic result
```

6. Past Exam Questions

6.1 Summer 1998

(5) 30/3/62

4. Use the *Moment-area method (Mohr's Theorems)* to determine, for the frame shown in Fig. Q4.

(*a*) the bending moment in the frame and

(*b*) the deflection at D.

Sketch the bending moment diagram giving the value of the bending moment at all salient points.

Sketch the deflected shape of the frame.

Take EI = 4000 kNm^2 (25 marks)

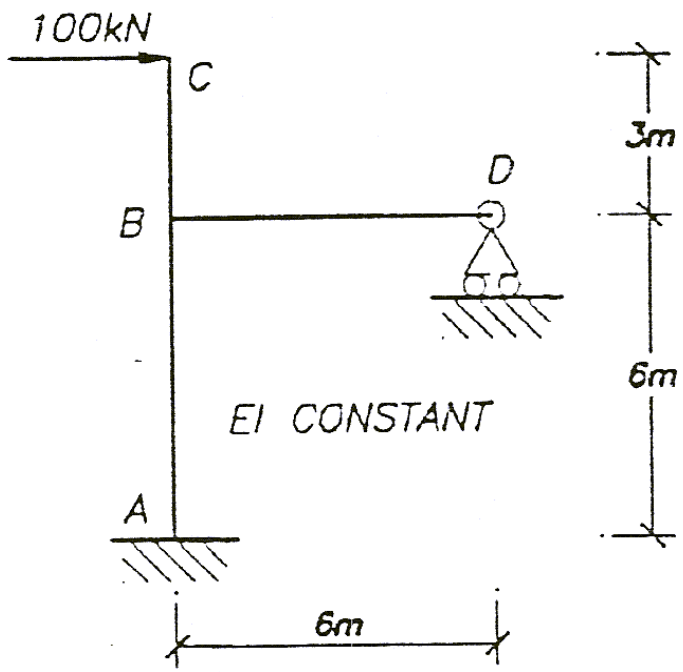


FIG Q4

Ans. $V_C = 75\text{kN}\uparrow$, $\delta_{Dx} = 4500/EI \rightarrow$

6.2 Summer 2005

Using the ***Moment-Area Method (Mohr's Theorems)***:

(a) Determine the bending moments in the frame shown in Fig. Q3 when a load of 90kN is applied at B as shown.

Hence sketch the bending moment diagram.

(15 marks)

(b) Calculate the horizontal deflection of joint C .

(10 marks)

Assume E = 200 kN/mm².

Assume I = 8×10^7 mm^4.

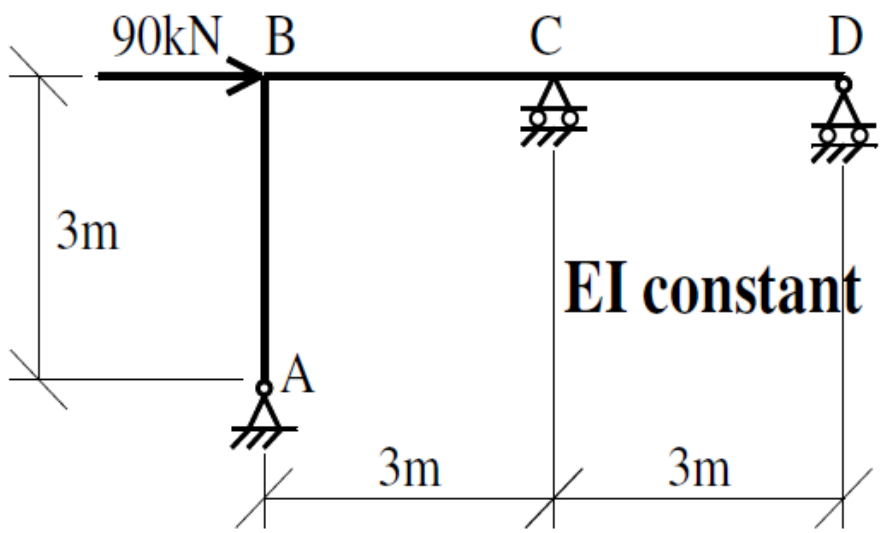


FIG. Q3

Ans. $V_D = 22.5\text{kN} \downarrow$, $\delta_{Dx} = 1519/EI \rightarrow$

6.3 Summer 2006

Use the ***Moment-area method (Mohr's Theorems)*** to determine, for the beam shown in Fig. Q4,

(a) the deflection at hinge C and

(b) the maximum deflection in span AB.

Sketch the deflected shape of the beam.

Take EI = 4000 kNm² .

(25 marks)

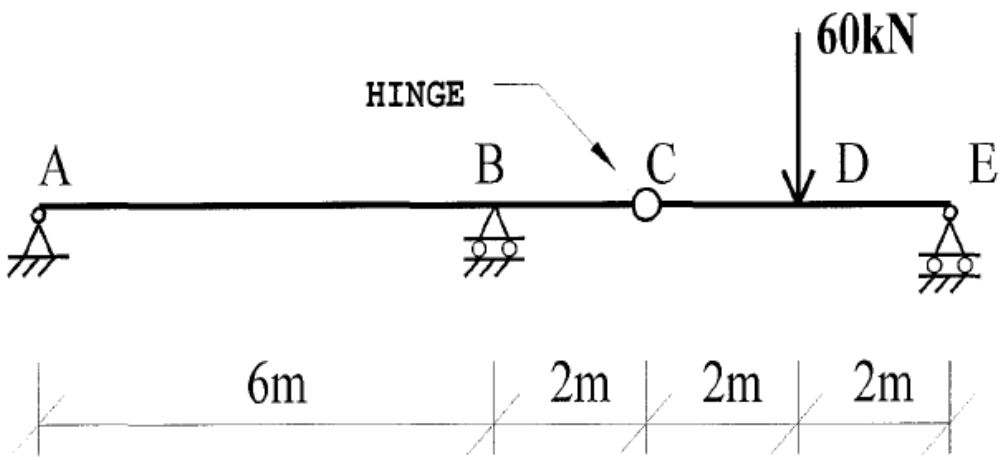


FIG. Q3

Ans. $\delta_C = 36\text{ mm}\downarrow$, $\delta_{\max|AB|} = 27.6\text{ mm}\uparrow$

6.4 Summer 2007

4. (a) For the frame shown in Fig. Q4(a), using *Mohr's Theorems*:

(i) Determine the vertical reaction at joint *C*;

(ii) Draw the bending moment diagram;

(iii) Determine the horizontal deflection of joint *C*.

Note:

You may neglect axial effects in the members.

Take $EI = 36 \times 10^3$ kNm^2 for all members.

(*15 marks*)

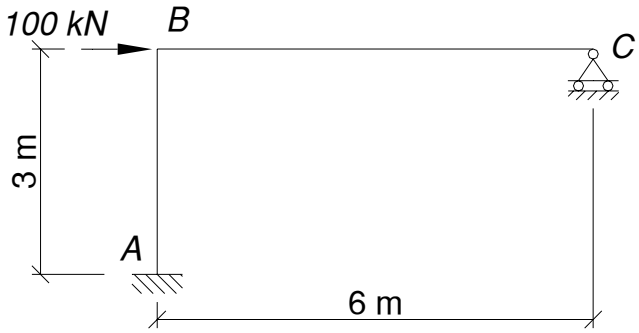


FIG. Q4(a)

Ans. $V_C = 15\text{kN} \uparrow$, $\delta_{Cx} = 495/EI \rightarrow$

6.5 Semester 1 2007/8

QUESTION 3

For the beam shown in Fig. Q3, using the *Moment-Area Method* (*Mohr's Theorems*):

(i) Draw the bending moment diagram;

(ii) Determine the maximum deflection;

(iii) Draw the deflected shape diagram.

Note:
Take $EI = 20 \times 10^3 \text{ kNm}^2$.

(40 marks)

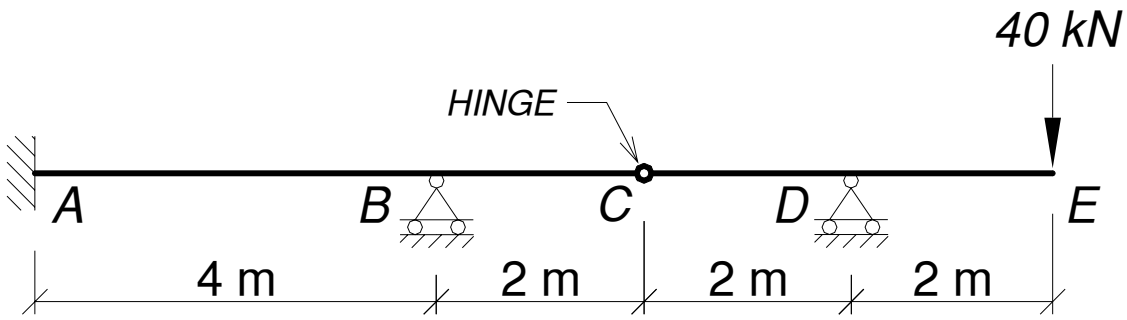


FIG. Q3

Ans. $V_C = 70\,\text{kN}\downarrow$, $\delta_{\max,AC} = 47.4/EI$, $\delta_C = 267/EI$, $\delta_E = 481/EI$

6.6 Semester 1 2008/9

QUESTION 3

For the frame shown in Fig. Q3, using the *Moment-Area Method* (*Mohr's Theorems*):

(iv) Draw the bending moment diagram;

(v) Determine the vertical and horizontal deflection of joint *E*;

(vi) Draw the deflected shape diagram.

Note:
Take $EI = 20 \times 10^3 \text{ kNm}^2$.

(40 marks)

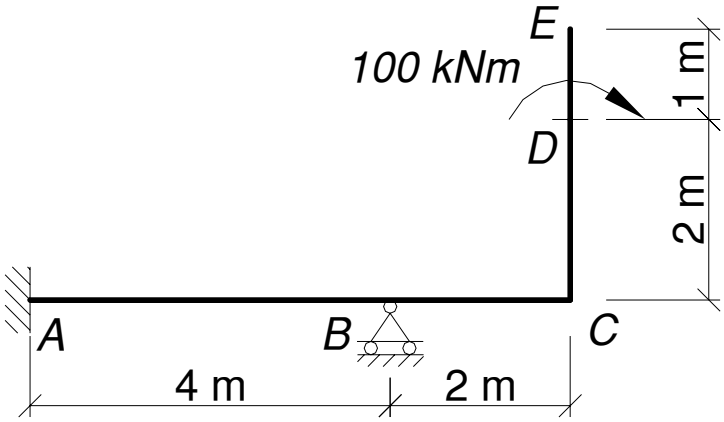
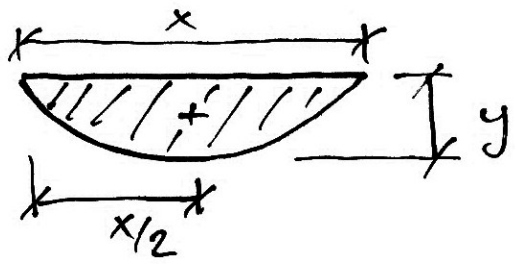
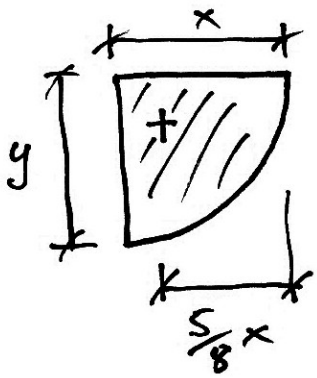
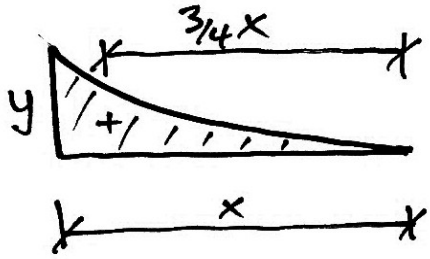


FIG. Q3

Ans. $V_B = 37.5\text{kN}\uparrow$, $\delta_{Ey} = 20 \text{ mm}\downarrow$, $\delta_{Ex} = 65 \text{ mm}\rightarrow$

7. Area Properties

These are well known for triangular and rectangular areas. For parabolic areas we have:

Shape	Area	Centroid
	$A = \frac{2}{3}xy$	$\bar{x} = \frac{1}{2}x$
	$A = \frac{2}{3}xy$	$\bar{x} = \frac{5}{8}x$
	$A = \frac{1}{3}xy$	$\bar{x} = \frac{3}{4}x$